



# Microstructure formation during gas flow-assisted additive manufacturing of a metallic glass powder on ground and in microgravity

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## Abstract

We studied bulk metallic glasses produced from gas flow-assisted laser-based powder bed fusion process, which is capable of additive manufacturing metallic parts in microgravity. A Zr-based bulk metallic glass composition  $\text{Zr}_{59.3}\text{Cu}_{28.8}\text{Al}_{10.4}\text{Nb}_{1.5}$  has been processed on ground and in microgravity in a compact sounding rocket payload MARS-M. Microstructure characterization was performed using electron microscopy and X-ray diffraction computed tomography, which cope with small amounts of sample materials, especially for those fabricated under microgravity conditions. Very similar microstructures and crystalline fractions are observed in sample manufactured on ground and in microgravity, which shows that process parameters of conventional laser powder bed fusion for manufacturing metallic glasses can be transferred to the processes in microgravity. Two different origins of crystallization have been identified in the  $\text{Zr}_{59.3}\text{Cu}_{28.8}\text{Al}_{10.4}\text{Nb}_{1.5}$  sample. The preferred occurrence of  $\text{CuZr}_2$  at the interlayer boundaries is likely a result of recrystallization from the undercooled melt and hence associated with laser scanning strategy. In contrast, the more uniformly distributed  $\text{Al}_3\text{Zr}_4$  phase is considered to be triggered by the formation of  $\text{Cu}_2\text{Zr}_4\text{O}$ . Thus, for the fabrication of fully amorphous builds both on ground and in space, our findings point to higher scanning speeds and lower oxygen contents, while the latter can also be used to tune the crystalline fractions in the sample.

**Keywords** Gas flow-assisted laser-based powder bed fusion · Microgravity · Glass-forming alloys · X-ray diffraction tomography

## 1 Introduction

Bulk metallic glasses (BMGs) are a relatively late arrival in the history of materials science, coming to existence in the early 1960s. Due to their amorphous structure and hence having no grain boundaries, they can display excellent corrosion resistance, good mechanical properties, and low friction coefficient, making them attractive for both ground and space applications [1, 2].

Generally produced by casting, the size of BMGs is usually limited to a few millimeters to centimeters in thickness due to dropping of cooling rates and therefore increased probabilities of crystallization when dimensions are increased during casting [3]. Optimization of the critical

casting thickness of fully glassy samples, also known as glass forming ability (GFA), can be only achieved by delicate adjustment of the alloy composition. Recently, however, these materials have begun to be used in additive manufacturing (AM). By building layer by layer, it is possible to circumvent the usual size limitations and to form an amorphous part larger than that attained by casting [4–6].

Of all AM processes, laser-based powder bed fusion (LPBF) is one of the most versatile techniques due its range of possible geometries and scalable process parameters. It can be adapted to a wide range of materials [7–9], such as metals [4, 6, 10, 11], ceramics [12], and polymers [13, 14]. This makes it an attractive process for materials research as well as for industry. By now, this technology has become a reliable alternative for building structural parts [15] and parts with complex geometry or from materials difficult to handle in traditional ways [16].

Extended author information available on the last page of the article



AM technologies are considered as a necessity for space applications, particularly for exploration missions, because they contribute to saving material and reducing mass to transport as well as production time, allowing a fast and flexible response to the loss or damage of parts. However, the largest challenge in manufacturing parts in an LPBF process independent from gravity is the powder handling. On ground, the powder deposition relies on stabilization by gravity. It is, hence, necessary to compensate for lacking gravitational forces to use a powder bed under space conditions. One possibility is to use a gas flow-assisted powder deposition developed by Zocca et al. [17, 18], in which a porous build-platform is used in combination with a vacuum pump-driven reduced pressure. This creates a gas flow through the build-platform which stabilizes the powder. We have recently demonstrated that it is possible to produce BMG samples based on this gas flow-assisted powder handling system, using a compact additive manufacturing device MARS-M to perform LPBF onboard the DLR's MAPHEUS sounding rocket [19].

The combination of the gas flow-assisted powder deposition process and glass forming alloys introduces several aspects that differ from the conventional LPBF processes. The presence of a continuous gas flow might affect, e.g., heat transfer, pore formation, or impurities in the atmosphere. Bulk metallic glass forming alloy melts exhibit slow liquid dynamics, i.e., higher viscosity, more sluggish mass transport compared to that of the casting alloys. The sounding rocket payload is a Cartesian printer, which provide lower scanning speed compared to commercial printer on ground. Nevertheless, this means also prolonged time for phase and defect formations, particularly in the case of BMGs, which allows their origin to be studied in more details.

Therefore, in the following, we present a detailed examination of the parts produced in lab and in microgravity using the gas flow-assisted LPBF process for a BMG alloy composition widely used for AM, to understand mechanisms that control the microstructure formation during the process, as well as to evaluate the application perspective of the process on ground and under microgravity. The compositions studied here,  $\text{Zr}_{59.3}\text{Cu}_{28.8}\text{Al}_{10.4}\text{Nb}_{1.5}$ , have been considered as a possible material dedicated for efficient excavating tools during space exploration missions, owing to the excellent mechanical properties of BMGs, for which the relevant dimensions can be only manufactured additively [20]. The knowledge on these processes, in particular whether and how gravity affects the AM, allows to identify technologies with suitable materials and process conditions tailored for these applications.

Because the currently microgravity time available is limited, only a few layers can be produced under these conditions. Consequently, characterization techniques requiring only small amounts of material were chosen: scanning

electron microscopy (SEM), transmission electron microscopy (TEM), as well as X-ray diffraction computed tomography at a synchrotron light source. With these, we show that gravity mainly affects the morphology of the print at large scale, on the order of millimeters, but not the microstructure and phase formation. In addition, with the spatial resolved diffraction technique, different origins of the crystalline phases have been identified.

## 2 Methods

### 2.1 Materials/alloys/powders

Powders of the metallic glass forming alloy AMLOY-Zr01 used in the current study ( $\text{Zr}_{59.3}\text{Cu}_{28.8}\text{Al}_{10.4}\text{Nb}_{1.5}$  at.%, previously known as AMZ4) were purchased from Heraeus GmbH. The alloy is developed using a commercial grade Zr705 alloy, containing Nb, among other impurities, and is typically cheaper [21]. This also leads to an oxygen content of about 1 at.% in the alloy. The melt properties and crystallization behavior have been well studied. In particular, it has been shown that the amount of oxygen present does not affect the melt viscosity significantly, but lowers the degrees of undercooling and GFA [21, 22]. The powder was used at 15 – 45  $\mu\text{m}$  particle size, produced by inert gas atomization [23] and is amorphous, as determined by X-ray diffraction analysis (see Appendix A).

### 2.2 Additive manufacturing

All the additively manufactured samples were processed in the MARS-M sounding rocket payload [19]. A laser of 976 nm was used, with a maximum power of 283 W. The laser spot size is of about 80  $\mu\text{m}$ . All samples were produced under (circulating) Ar atmosphere, with a residual  $\text{O}_2$  content below 0.45%. The single-layer thickness was set to 100  $\mu\text{m}$ . Samples were built on porous SS316L stainless steel build-platforms<sup>1</sup>. An overview of the lab and  $\mu\text{g}$ -samples investigated is listed in Table 1.

#### 2.2.1 Sample printed on ground

A “twinline” scanning strategy was used on ground, which involves two alternating layers being displaced by half a hatch distance to each other [19]. This was found to increase density within the part (as determined by SEM observations). The scanning direction is also inverted for each new layer, as this was found to reduce residual stress in the built,

<sup>1</sup> AISI 316 L/B, nominal composition: Cr 16.0–18.0 wt.%, Ni 10.0–14.0 wt.%, C  $\leq$  0.03 wt.%, Mo 2.0–3.0 wt.%, Fe balanced.



**Table 1** Overview of the parameters used for the samples characterized in the current study

sample	laser power (W)	scanning speed (mm/min)	layers
Lab (parameter studies)	80	4000	20
	80	4750	20
	80	5500	20
	115	5500	20
	135	4000	20
	170	4000	20
	170	4750	20
	170	5500	20
flight sample	75	4000	8
lab equivalent	75	4000	8

and hence preventing detaching of the part from the platform. The ground samples fabricated are 20 layers high, with a layer thickness of 100  $\mu\text{m}$  and a hatch distance of 300  $\mu\text{m}$ , which gives a sample cross-section of  $\sim 2 \times 2 \text{ mm}^2$  (perpendicular to the laser scan direction). The laser power used covers a range from 55 to 195 W, while the scanning speed varies between 4000 and 5500  $\text{mm min}^{-1}$  (a sample grid was manufactured with different laser power and scanning speeds, see Fig. 4 in Ref. [19]). The chosen range of parameters does not span the entire possibilities of the device, but is a reduced range within which the resulting samples can be processed for investigation (i.e., the powder is fully melted and the samples are sufficiently dense and do not shatter upon removal from the platform). The quality of these samples (density and crystallinity) was then investigated by SEM and X-ray diffraction analysis to determine the optimal applicable parameters.

### 2.2.2 Additive manufacturing under microgravity

Samples manufactured under microgravity conditions were processed in the MARS-M sounding rocket payload [19]. The MAPHEUS sounding rocket flight provides 6.5 min of microgravity. Thus, a sample with eight layers was fabricated, which provides enough material for analysis (see Fig. 1). It was built at 75 W and 4000  $\text{mm min}^{-1}$ . The geometry is made of thin lines (single slugs) because of the requirements for the gas flow-assisted powder deposition: when a new layer is spread over the partly sintered powder bed, a low enough thickness of individual lines ensures that the powder is resting on top of it is maintained by the gas flow [24]. To study the influence of  $\mu\text{g}$  on the LPBF process, a lab equivalent sample was also built under the same conditions (same built geometry, process parameters, and built height).



**Fig. 1** View of the flight sample built during the MAPHEUS-10 sounding rocket flight. The thickness of the sample is limited to eight-layers due to the available microgravity time of 6.5 min. The build is completely welded to the platform without detaching. The sample thickness mildly decreases toward the right half, due to feed-stock particle transport coming in from the left

### 2.3 Preparation steps for sample characterization

After building, the twinline samples processed on ground were either mechanically removed with pliers or left on the build-platform and cut into sections while still attached to the platform (as shown in Fig. 6). If the samples were meant for microscopy, they were then embedded in epoxy. The embedded samples were then ground and polished (final finishing with Buehler MicroCut<sup>®</sup> P4000, ChemoMet<sup>®</sup> polishing cloth, and MasterPrep<sup>™</sup> suspension) to a mirror-like surface and examined using both optical microscopy and SEM. The samples used in the X-ray diffraction experiments were detached from the build-platform but not embedded.

In opposition to the ground samples, the  $\mu\text{g}$ -sample and its lab equivalent were left as-built as—due to the small thickness as well as the porosity of the build-platform— attempts to remove such samples from their platform proved detrimental to the integrity of the sample. The samples for Transmission Electron Microscopy (TEM) were prepared from the AMLOY-Zr01  $\mu\text{g}$ -sample and its lab equivalent by means of Focused Ion Beam (FIB) using an FEI Strata 400 with Ga liquid metal ion source as well as a HELIOS dual beam with Xe plasma source. The macroscopic sampling site is the same single built-up segment of reticular structure marked by white arrows in Fig. 9. The two segments were cut free from the rest of the structure, followed by the preparation of an FIB lamella from each. The region of interest for FIB lift-out was located close to the interface between two layers at about half the height of the built-up structure. The dimensions of the FIB lamellae are  $\sim 17 \times 4\text{--}7 \text{ }\mu\text{m}$  (height  $\times$  depth).



## 2.4 Electron microscopy

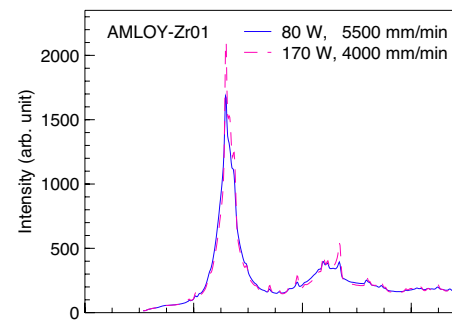
Scanning electron microscopy (SEM) was performed on a Zeiss Merlin Scanning Electron Microscope. The backscattered electron detector was used for all figures shown with energies ranging from 7 to 20 kV and a probe intensity of 2 nA. Energy-Dispersive X-ray Spectroscopy (EDS, Oxford Instruments) was performed for compositional analysis, while Electron Backscatter Diffraction (EBSD, Oxford Instruments) was performed for crystallographic analysis.

The near-interface microstructure of AMLOY-Zr01 samples was investigated by scanning transmission electron microscopy (S)TEM in bright-field (BF) mode using a JEOL JEM-F200 device operated at an acceleration voltage of 200 kV and equipped with a GATAN OneView camera for high-resolution imaging. The amorphousness was investigated by Selected Area Electron Diffraction (SAED) and high-resolution (HRTEM) imaging. HRTEM images of phases were processed by FFT and inverse FFT using ImageJ [25]. The resulting SAED-like diffraction patterns and lattice plane fringes enabled the identification of the phases. Additional information on the chemical composition of the phases was obtained in STEM mode by EDS using an Oxford Instruments detector. For both electron microscopy techniques, the preparation steps of the sample can be found in Sect. 2.3.

## 2.5 X-ray diffraction experiments

Both the X-ray micro-diffraction and the diffraction computed tomography (XRD-CT) focus primarily on the samples produced on ground, which have larger dimensions (cross-sections of about  $2 \times 2 \text{ mm}^2$  as mentioned above). X-ray micro-diffraction as well as diffraction tomography measurements were performed at the second experimental hut (EH2) of the P07 High Energy Materials Science Beamline at DESY (Deutsches Elektronen-Synchrotron, Hamburg, Germany). For the micro-diffraction experiment scans were performed in transmission on detached ground samples along the build direction across a cross-section of  $2 \times 2 \text{ mm}^2$ . An incoming X-ray energy of 103.5 keV was used with a beam size of  $2 \text{ }\mu\text{m}$  (vertical)  $\times$   $30 \text{ }\mu\text{m}$  (horizontal). A DECTRIS PILATUS3 X CdTe 2 M area detector was placed at 348 mm downstream from the sample position, which gives an accessible momentum transfer  $q$ -range of  $1 - 15 \text{ }\text{\AA}^{-1}$ . The measurements were done in continuous scanning mode (at 2 s), with a sampling rate of 0.25 s exposure time per diffraction image (4 Hz), resulting in a covered scan range of  $0.5 \text{ }\mu\text{m}$  per image, and a spatial resolution on the sample of about  $2.5 \text{ }\mu\text{m}$ .

Figure 2 shows the average pattern over a whole sample for the samples with the highest and the lowest energy



**Fig. 2** Averaged X-ray micro-diffraction patterns of samples built with the lowest and highest energy density for AMLOY-Zr01

density as an example, confirming the crystalline phases visible in the X-ray micro-diffraction measurements.

In the case of XRD-CT measurements, the photon beam energy was reduced to 73.3 keV to have higher flux and detection efficiency. The beam size was  $2 \text{ }\mu\text{m} \times 30 \text{ }\mu\text{m}$ . A DECTRIS EIGER2 X CdTe 4 M detector was placed at a distance of 400 mm covering a usable  $q$ -range of  $0.5 - 8 \text{ }\text{\AA}^{-1}$  running at 500 Hz (2 ms exposure time). The reconstructed datasets consist of 800 projections distributed over an angle of 360 deg. For each projection, a range of 3.6 mm was scanned across the sample in the direction perpendicular to the incoming beam with a constant speed of  $1 \text{ mm s}^{-1}$ . The resulting voxel size is  $2 \text{ }\mu\text{m} \times 2 \text{ }\mu\text{m}$ .

The obtained 2D diffraction patterns were integrated to obtain 1D intensity profile using the pyFAI (Fast Azimuthal Integration using Python) software package [26]. For both experiments, integration parameters were obtained using  $\text{CeO}_2$  calibrants. In the case of XRD-CT, the reconstruction was performed using a Matlab code developed at DESY.

## 2.6 Differential scanning calorimetry

Calorimetric experiments were performed using a power-compensated DSC8500 (PerkinElmer, USA). After equilibration at 303 K for 2 min, each sample was heated twice from 303 K to 853 K at heating rates of  $0.33 \text{ K s}^{-1}$ . The second run corresponds to the crystalline baseline, which was later subtracted from the first scan. The crystallization enthalpy was then determined by integrating the crystallization peak. The crystallization enthalpy release of casted, fully amorphous materials of AMLOY-Zr01 was used as a reference to estimate the amorphous fraction of the printed samples. The fully amorphous sample was produced by alloying high-purity elements (Zr, Cu, Al, Nb) in an arc-melter under high-purity Ar atmosphere, and subsequently suction casted into water-cooled copper mold. The casting dimensions were kept below the critical casting thickness. The amorphous nature of the casted sample was confirmed by XRD.



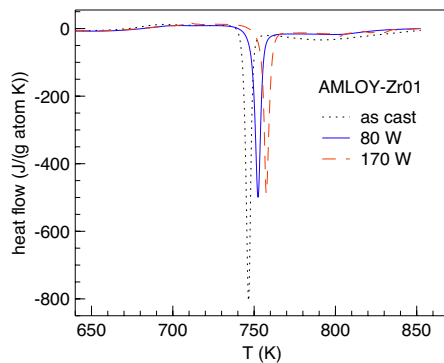
### 3 Results and discussion

#### 3.1 Crystallinity

As already shown in the previous publication, the obtained samples were not fully amorphous [19], as can be also seen in Fig. 2. Compared to the additively manufactured AMLOY-Zr01 samples in the literature where fully amorphous print is achievable, the main reason for the partial crystallinity of the sample built in the MARS-M device is the limited scanning speed. The speeds recommended in the literature for AMLOY-Zr01 at  $2000\text{ mm s}^{-1}$  are at least an order of magnitude higher than what is attainable with MARS-M [23, 27, 28]. This is due to the mechanical and size requirements to the device for it to survive several rocket starts, re-entries, and landings. In view of long-term space applications, once the acceleration at launch will be reduced, higher scanning speeds are realizable without the necessity of a Cartesian setup.

Figure 3 shows the DSC scan curves of the printed and casted AMLOY-Zr01 samples. Samples were produced using two different laser powers: 80 W and 170 W. The laser scanning speed was kept constant at  $4750\text{ mm min}^{-1}$ . Glass transition can be recognized as a small endothermic event, whereas crystallization events can be identified as large exothermic peaks. The characteristic temperatures and the obtained crystallization enthalpy are listed in Table 2.

The differential scanning calorimetry reveals a crystallization enthalpy of  $3350\text{ J mol}^{-1}\text{K}^{-1}$  for the AMLOY-Zr01 sample printed with 80 W,  $4750\text{ mm min}^{-1}$ , and  $3020\text{ J mol}^{-1}\text{K}^{-1}$  for that printed with 170W,  $4750\text{ mm min}^{-1}$ . This corresponds to an amorphous fraction of 69% and 62%, respectively, where the measured crystallization enthalpy of fully amorphous  $\text{Zr}_{59.3}\text{Cu}_{28.8}\text{Al}_{10.4}\text{Nb}_{1.5}$  sample



**Fig. 3** DSC curves of as casted and printed samples of AMLOY-Zr01. For the printed samples, two different laser powers were used: 80 W and 170 W for each alloy. A scanning speed of  $4750\text{ mm min}^{-1}$  was kept same for all samples, which also corresponds to the one used in the microgravity experiment. Positive values on the y-axis represent endothermic heat flow

**Table 2** Characteristic temperatures and the crystallization enthalpy determined in the DSC measurements for the casted and printed AMLOY-Zr01 alloys

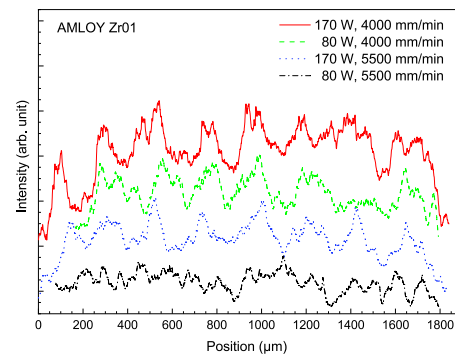
Sample	$T_g$ (K)	$T_x$ (K)	$\Delta H$ ( $\text{Jg}^{-1}$ )
AMLOY-Zr01			
as cast	661.9	741.0	63.37
80 W	664.7	745.8	43.73
170 W	663.3	751.3	39.43

is  $4854\text{ J mol}^{-1}\text{K}^{-1}$ . It seems that lower laser power at the same scanning speed results in less crystallization.

#### 3.2 Distribution of amorphous and crystalline fractions

When looking at the variation in integrated peak intensity of the main Bragg peaks, a periodic repetition of the intensity maximum and minimum is observed, as shown in Fig. 4 for an AMLOY-Zr01 sample produced at four different process parameters (80 W, 4000 and  $5500\text{ mm min}^{-1}$  as well as 170 W, 4000 and  $5500\text{ mm min}^{-1}$ ). An approximate  $200\text{ }\mu\text{m}$  periodicity is obtained. On the crystallized AMLOY-Zr01 after annealing measured in the same micro-diffraction setup, this periodicity was no longer visible. Thus, this effect is not to be traced back to minor changes in sample thickness, but indeed from changes in the ratio between the crystalline and the amorphous fractions at different build height in the sample.

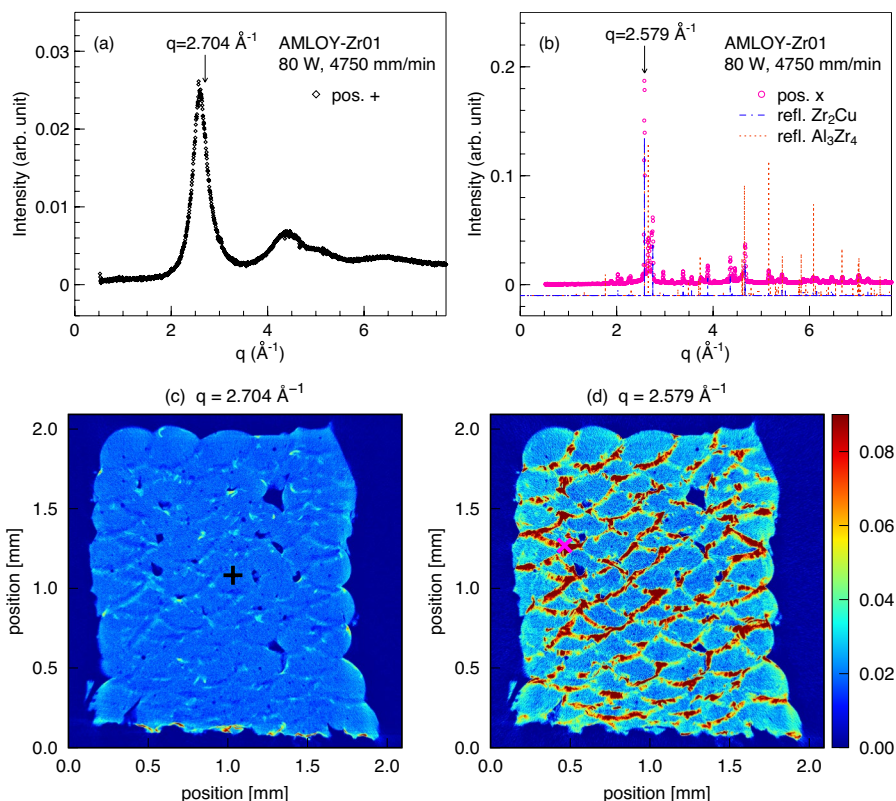
This kind of periodicity in the patterns was observed for all the samples except the lowest energy density (i.e., 80 W,  $5500\text{ mm min}^{-1}$ ). Also, as shown in Fig. 2 and 4, the diffraction pattern and the integrated intensity do not show qualitative differences. Thus, it can be concluded that in process parameter range (laser power 80–170W, scan speed



**Fig. 4** Variation in the integrated intensity of the main Bragg peak for AMLOY-Zr01 ( $q$ -range between  $2.56$  and  $2.64\text{ }\text{\AA}^{-1}$ ) as the beam scans down the samples built at different processing parameters



**Fig. 5** XRD-CT of cross-sections of AMLOY-Zr01 built with  $4750 \text{ mm min}^{-1}$  and  $80 \text{ W}$ : (a) and (b) diffraction spectra at the positions shown in (c) and (d), respectively. Spectra are averaged over 9 pixels around the central position. (c) and (d) Intensity distribution over the entire sample at  $q$ -values corresponding to amorphous  $q = 2.704 \text{ \AA}^{-1}$  and crystalline material  $q = 2.579 \text{ \AA}^{-1}$ . The brighter parts at the bottom of the sample in (c) stem from mixing with the steel platform, as shown later in Fig. 6. Intensity scales are kept same for both graphics



$4000 - 5500 \text{ mm min}^{-1}$ ), the crystalline phases formed do not show significant changes, at least for those measurable in the X-ray diffraction experiment. The observed intensity variation in Fig. 4 can be confirmed by SEM micrographs reported previously [19]. The periodicity would correspond to that of every two built layers, which can be understood in light of the developed scanning strategy, because odd-numbered layers are displaced by half a hatch distance (i.e.,  $150 \text{ }\mu\text{m}$ ) compared to even-numbered ones. However, both in X-ray micro-diffraction and in the SEM observations, also deviation from the  $200 \text{ }\mu\text{m}$  periodicity has been identified, indicating non-regular distribution of the crystalline fraction and layer boundaries in the sample.

As XRD-CT provides spatially resolved diffraction information, it is better suited to view these partially crystalline regions (non-destructively). Figure 5 shows the XRD-CT cross-sections and the corresponding diffraction patterns of the AMLOY-Zr01 sample built with a scanning speed of  $4750 \text{ mm min}^{-1}$  and a laser power of  $80 \text{ W}$ . For each pixel ( $2 \text{ }\mu\text{m}$  size), full diffraction information is available, as shown in Fig. 5a and b. For two  $q$  values  $2.704 \text{ \AA}^{-1}$  and  $2.579 \text{ \AA}^{-1}$ , intensity maps of the sample are shown in Fig. 5c and d. These two  $q$ s are chosen to resolve the spatial distribution of the amorphous and the crystalline phases:  $q = 2.704 \text{ \AA}^{-1}$  represents the amorphous region and is slightly above the position of the first structure factor

maximum of the glass;  $q = 2.579 \text{ \AA}^{-1}$  corresponds to (103) and (111) Bragg reflections of the  $\text{CuZr}_2$  and the  $\text{Al}_3\text{Zr}_4$  phases, respectively<sup>2</sup>, which are also later identified by SEM (Fig. 7), using particularly EBSD.

For the crystalline phases, the choice of the Bragg reflections is based on the consistently reported  $\text{CuZr}_2$  and  $\text{Al}_3\text{Zr}_4$  phases as the main crystallization products for additively manufactured AMLOY-Zr01 in the literature [29–34], with the presence of some  $\text{Cu}_2\text{Zr}_4\text{O}$  fractions. However, this choice of the representation does not exclude the potential presence of other crystalline phases in the sample. Nevertheless, identifying minor crystallization products is either beyond the capability of the spatial resolution of the XRD-CT configuration used here, or the positions of the Bragg peaks is rather close to those of the  $\text{CuZr}_2$  and  $\text{Al}_3\text{Zr}_4$  phases. The presence of other crystalline phases will be discussed together with the transmission electron microscopy results.

It can be seen that the intensity distribution is much more uniform in Fig. 5c (amorphous) compared to that in Fig. 5d (crystalline). This shows that the amorphous phase is present through out the sample, whereas the crystalline phases

<sup>2</sup> Obtained by crystal structure analysis (Le Bail method using following crystalline structures:  $\text{CuZr}_2$   $I4/mmm$  with  $a = 3.2315 \text{ \AA}$  and  $c = 11.1597 \text{ \AA}$ ,  $\text{Al}_3\text{Zr}_4$   $P6/mmm$  with  $a = 5.4648 \text{ \AA}$  and  $c = 5.4051 \text{ \AA}$



mainly appear in the regions between the layers. The curvature observed between the layers shows the contact geometry of the melt with the previous solidified track. The thickness of these boundary layers is estimated to be few tens of microns. The results confirm that the fraction of the crystalline phases is minor in the sample, as well as the observed irregularity of the  $\sim 200\ \mu\text{m}$  periodicity in Fig. 4, as these interlayer boundaries are not aligned entirely parallel.

### 3.3 Influence of build-platform

In the MARS-M facility, the samples are manufactured onto a sintered porous SS316L stainless steel build-platform, and the first layer of the build is very much embedded and welded in this platform [19]. A cross-section of an as-built AMLOY-Zr01 sample ( $5500\ \text{mm min}^{-1}$  and  $170\ \text{W}$ ) on the porous steel build-platform is shown in the left panel of Fig. 6.

When looking more closely at the sample-platform interface, mixing is visible, as shown in the right panel of Fig. 6. EDS analysis revealed some chemical mixing (see Table 3) and potentially formation of compositions, so far unidentified.

Since during processing of at least the first layer, the alloy melt is in contact with the platform, such a formation of new (e.g., iron containing) alloys can be expected. However, X-ray diffraction measurements reveal that such physical or chemical mixing is localized within the first, maximum the second layer (see diffraction pattern in Appendix Fig. B.1). It can also be seen from the XRD-CT images (Fig. 5c) that only a thin layer of a few tens of micron thickness at the bottom of the built sample shows a different diffraction pattern. This allows us to draw the conclusions that the stainless steel affects only a very limited region of the sample, and even in the flight samples with less layers, the majority part of the printed material has the same composition as of the powder.

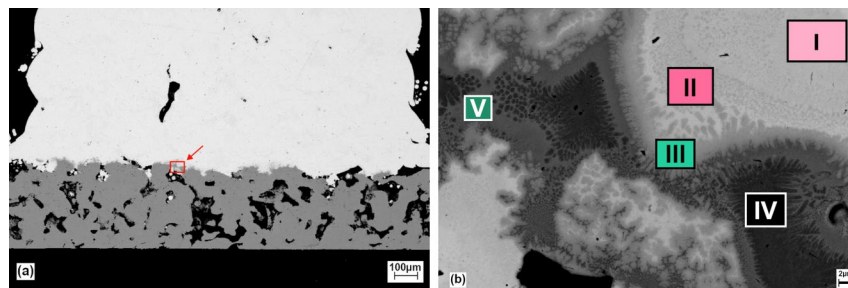
### 3.4 Details and origin of the crystalline phases

In the following section, the details of the crystalline phases in the printed sample will be presented.

Figure 7a shows an SEM image of the crystalline regions of one powder layer including the boundaries areas to the upper and lower layers. It appears that the crystallization is of two types/steps. First, most of the crystal grains (dark areas in the image) are located in the interlayer region. These are known as the so-called heat-affected zones (HAZ) discussed in the literature [35–40], particularly from single track melting experiments on the AMLOY-Zr01 alloy [41]. This is a result of crystallization from undercooled melt where close to the laser heated area, the glassy materials was heated above the glass transition. Second, some of the crystal grains have been displaced farther away from the layer interface. In this case, a more curvy morphology of the crystals can be observed. This indicates that these crystalline fractions are likely formed in the melt pool where convective flows are present, leading the crystals being displaced during or after formation.

**Table 3** Measured concentrations in at.% at different points of Fig. 6. The nominal composition of the alloy is  $\text{Zr}_{59.3}\text{Cu}_{28.8}\text{Al}_{10.4}\text{Nb}_{1.5}$ . Cr and Ni are sourced from the base plate

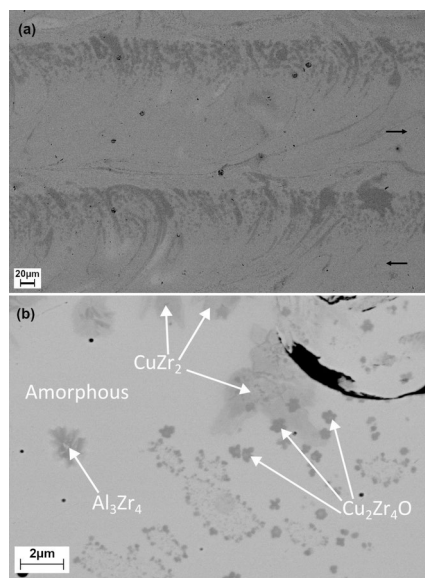
Element	I	II	III	IV	V
O	11.75	13.71	13.96	7.12	5.13
Al	7.03	6.33	3.40	1.29	0.38
Cr	0.79	1.58	7.56	14.34	18.57
Fe	3.94	10.31	36.43	55.27	65.67
Ni	1.27	3.26	5.67	9.77	8.82
Cu	24.38	22.37	7.66	3.19	0.65
Zr	48.42	40.33	24.08	8.84	0.79
Nb	2.42	2.11	1.26	0.17	0



**Fig. 6** (a) SEM micrograph of the cross-section of a “twinline” AMLOY-Zr01 sample as-built and still attached to the porous build-platform. The scale represents  $100\ \mu\text{m}$ . The arrow points to the location where a zoomed-in view was taken as shown in the right picture. (b) A closer look at the interface area, presenting different crystalline

regions of varying compositions. The compositions determined by energy-dispersive X-ray analysis for each rectangle are presented in Table 3. The back-scattered electron detector was used. The scale bar represents  $2\ \mu\text{m}$





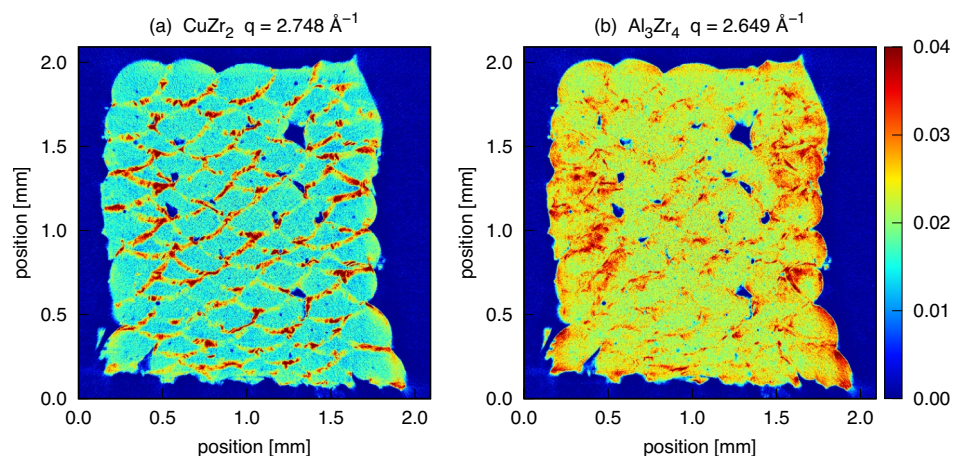
**Fig. 7** (a) Increased magnification of a cross-section of AMLOY-Zr01 built with  $4000 \text{ mm min}^{-1}$  and  $170 \text{ W}$ , parallel to the scanning direction. The scale represents  $20 \mu\text{m}$ . The arrows indicate the scan direction in a single layer. (b) Various crystals appearing in AMLOY-Zr01 samples in the darker (crystalline) areas. The scale represents  $2 \mu\text{m}$ . The crystals were identified thanks to EBSD

Figure 7b shows the different type of crystal phases identified by SEM employing both EDS and EBSD analysis in the melt pool region. In particular, EBSD analysis using the Crystallographic Analysis of the Lattice Metric (CALM) software [42] revealed the presence of tetragonal  $\text{CuZr}_2$ , hexagonal  $\text{Al}_3\text{Zr}_4$  and large cubic structure matched to  $\text{Cu}_2\text{Zr}_4\text{O}$  (CIF: 1220324, SG = 227,  $a = 12.28 \text{ \AA}$ ).<sup>3</sup> As

industrial grade AMLOY-Zr01 powder was used, and the process gases contained additional oxygen contents, the presence of the  $\text{Cu}_2\text{Zr}_4\text{O}$  phase is expected, which has been shown to be stable at an oxygen concentration greater than  $0.5 \text{ at.}\%$  [27]. The grain size of the observed  $\text{Cu}_2\text{Zr}_4\text{O}$  phase is typically below  $1 \mu\text{m}$ , as shown in Fig. 7b. The typical grain size of the  $\text{CuZr}_2$  and hexagonal  $\text{Al}_3\text{Zr}_4$  are larger, but are mostly around or below  $2 \mu\text{m}$ .

The observation of  $\text{CuZr}_2$  and  $\text{Al}_3\text{Zr}_4$  by SEM, particularly using EBSD, agrees with the XRD-CT results here. The  $\text{Cu}_2\text{Zr}_4\text{O}$  phase cannot be unambiguously identified in XRD-CT due to its small grain size and the overlapping Bragg reflections with other phases. The distributions of the  $\text{CuZr}_2$  and  $\text{Al}_3\text{Zr}_4$  in the bulk sample fabricated with  $80 \text{ W}$  laser power and  $4750 \text{ mm min}^{-1}$  scanning speed are shown by the XRD-CT results (Fig. 8). Here,  $q$  values of  $2.748 \text{ \AA}^{-1}$  and  $2.649 \text{ \AA}^{-1}$  are chosen, which correspond to the  $\text{CuZr}_2$  ((110) reflection) and the  $\text{Al}_3\text{Zr}_4$  phases ((200) and (102) reflections), respectively, where the Bragg peaks are more separated from each other. The higher intensity between the lines and layers in Fig. 8a indicates that the  $\text{CuZr}_2$  phase is more concentrated at the interface. In contrast,  $\text{Al}_3\text{Zr}_4$  crystals are distributed more in the volume, which leads to higher intensities in the bulk, as well as at the outer edge of the sample. Although it should be noted that the overall intensity is lower compared to that for  $\text{CuZr}_2$ , indicating at least qualitatively lower volume fraction of the  $\text{Al}_3\text{Zr}_4$  phase. Taking into account the considerably higher melting point of  $\text{Al}_3\text{Zr}_4$  compared to that of  $\text{CuZr}_2$ , it may be concluded that the  $\text{Al}_3\text{Zr}_4$  phase is preferentially nucleated at higher temperatures and hence also earlier in the melt pool.

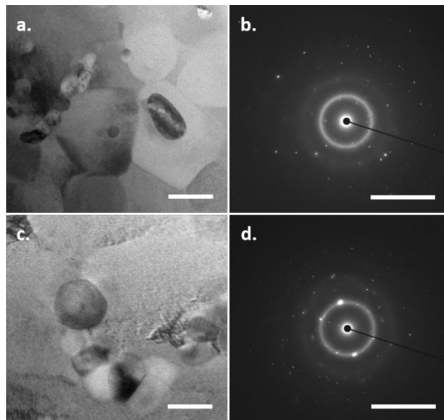
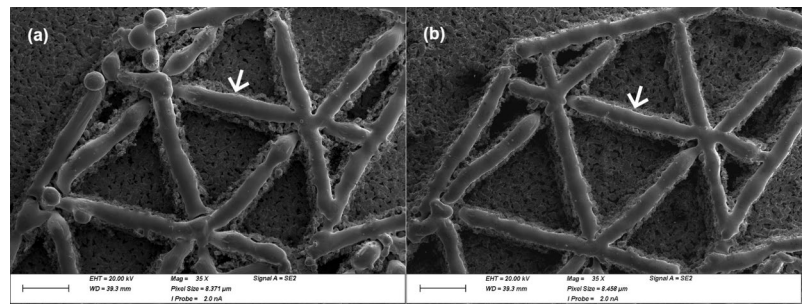
**Fig. 8** XRD-CT of cross-sections of AMLOY-Zr01 built with  $4750 \text{ mm min}^{-1}$  and  $80 \text{ W}$  at  $q$ -values corresponding to (a)  $\text{CuZr}_2$   $q = 2.748 \text{ \AA}^{-1}$  and (b)  $\text{Al}_3\text{Zr}_4$  crystalline phases  $q = 2.649 \text{ \AA}^{-1}$ . Intensity scales are the same for both graphics



<sup>3</sup> crystal structure data from Pearson crystallographic database were used [P. Villars, K. Cenzual, Pearson's Crystal Data: Crystal Structure Database for Inorganic Compounds (on DVD), Release 2020/2021, ASM International®, Materials Park, Ohio, USA]



**Fig. 9** SEM micrographs of the sample produced during the MAPHEUS-10 sounding rocket flight (a) and the equivalent lab sample (b). The background is the porous steel build-platform, as the samples were examined as-built. The scale represents 1 mm



**Fig. 10** TEM-BF images and the corresponding SAED patterns of MAPHEUS-10  $\mu\text{g}$ -sample (a.-b.) and lab equivalent (c.-d.) near the interface of two layers. Scale bar of BF images (a., c.) is 100 nm and of SAED patterns (b., d.)  $10\text{ nm}^{-1}$

### 3.5 Influence of microgravity on phase morphology

To study the influence of microgravity on the builds, two samples were built with the same manufacturing parameters and geometry—but differing gravity conditions. These samples were already introduced in a previous paper [19]. A scanning electron micrograph of each sample is shown in Fig. 9. The  $\mu\text{g}$ -sample presents more balling, which can be understood by the reduced hydrostatic pressure on the melted metals due to weightlessness, particularly at the ends of individual lines. The lab sample is smoother in appearance.

In terms of phase analysis, both samples were previously examined in the as-built condition in X-ray micro-diffraction [19]. From the diffraction results, no significant difference is observed in the diffraction patterns of both the  $\mu\text{g}$ - and lab sample. Because the X-ray diffraction experiment could not reveal more details in terms of higher resolution due to the limited thickness of the samples, additional investigations were performed employing TEM. TEM microstructure analysis obtained from the MAPHEUS-10  $\mu\text{g}$ -sample and the corresponding lab equivalent of AMLOY-Zr01 is shown in Fig. 10. The TEM samples were taken from the interlayer

**Table 4** Size of nanocrystalline spherical-shaped and ellipsoidal-shaped phases observed by TEM in  $\mu\text{g}$ -sample and lab equivalent of AMLOY-Zr01 in nm

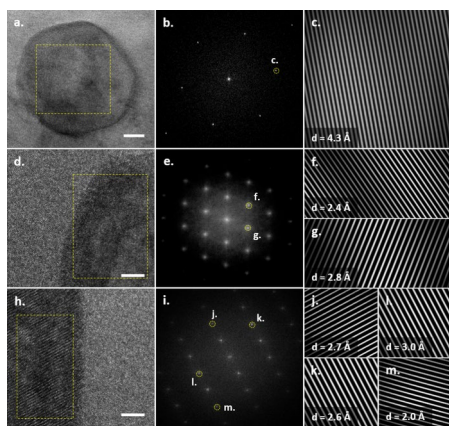
	Flight Sample (nm)	Lab Equivalent (nm)
Spherical	$21.12 \pm 6.10$	$94.67 \pm 25.62$
Ellipsoidal		
Length	$64.96 \pm 20.86$	$113.40 \pm 28.08$
Width	$33.70 \pm 10.50$	$68.28 \pm 16.86$
Aspect Ratio	$1.95 \pm 0.34$	$1.70 \pm 0.38$

regions between two building layers. Clusters of nanocrystalline phases with spherical or ellipsoidal morphology have been observed, as shown in Fig. 10a and Fig. 10c for the  $\mu\text{g}$  and lab sample, respectively. These clusters exhibit clearly defined phase boundaries to the amorphous matrix, as confirmed by the distinct diffraction spots next to the amorphous ring in the corresponding SAED patterns in Fig. 10b and Fig. 10d. No quasi-crystalline structures have been observed.

Average sizes of the nanocrystalline phases in this region are given in Table 4. Smaller sizes of the phases are measured in the  $\mu\text{g}$ -sample compared to the lab equivalent. This could be attributed to the fact that the outer temperature of the entire MARS-M payload at launch was low (below 300 K due to the weather conditions), so that at least part of the processing were performed during which the temperature of the base plate was also lower than that under laboratory conditions. However, considering the fact that TEM only samples a small fraction of the sample, this could be only one of the possible reasons, and we do not exclude other potential effects. Also, the base plate was warmed up by the heating laser as the AM proceeded. Thus, this impact is expected to diminish.

HRTEM-BF images, as shown in Fig. 11, were taken from the spherical and ellipsoidal particles in the  $\mu\text{g}$ -sample and in the lab equivalent. Figure 11a shows a HRTEM image of a spherical particle in the lab sample. The corresponding FFT and inverse FFT processed HRTEM images, shown in Fig. 11b and Fig. 11c, respectively, reveal an FCC structure with an average d-spacing of  $4.354 \pm 0.533\text{ \AA}$  (Fig. 11c). Since the zone axis of the





**Fig. 11** HRTEM-BF micrographs and corresponding FFT and IFFT processed images of the marked region of MAPHEUS-10 lab equivalent (a.-c.) and flight sample (d.-m) at the interface of two layers. The scale bar of HRTEM images is 20 nm (a.) and 5 nm (d.,h.)

pattern is clearly  $\bar{1}11$ , the diffraction spots close to it therefore belong to  $\{220\}$  plane family [43], and an average lattice parameter of  $12.284 \pm 0.152 \text{ \AA}$  is derived, which fits at best to  $\text{Cu}_2\text{Zr}_4\text{O}$  [27, 30, 44–47].

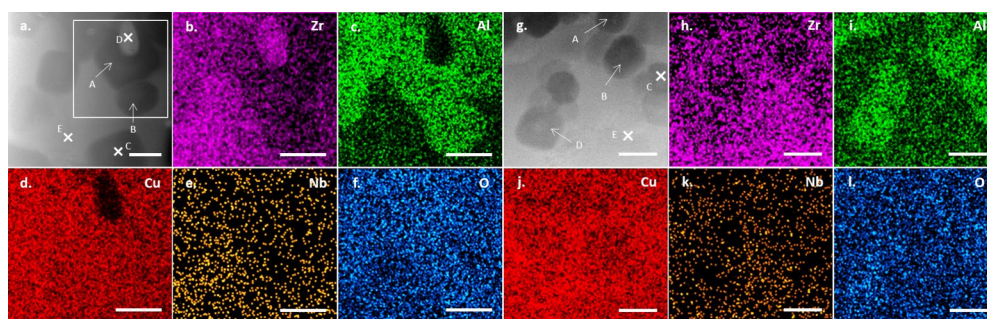
In the literature, a wide range of lattice parameters from  $11.9\text{--}12.28 \text{ \AA}$  is reported for the  $\text{Cu}_2\text{Zr}_4\text{O}$  phase [27, 30, 44–48], as a large amount of Al can be dissolved in  $\text{Cu}_2\text{Zr}_4\text{O}$  [34] and it is assumed that Al substitutes Zr in its lattice sites. Furthermore, it is also possible that Al substitutes Cu to some extent as proposed by Tidefelt et al., forming  $(\text{Al,Cu})\text{Zr}_2$  and  $(\text{Al,Cu})_3\text{Zr}_4$  phases [32]. Due to the larger atomic radius of Al compared to Cu, such substitution could lead to additional increases of the lattice constant [49]. With respect to the EDS maps in Fig. 12, the crystalline regions are enriched with Al and O. Therefore, it is reasonable that the derived lattice constant here is closer to the upper limit of the range reported in the literature [30, 46].

HRTEM images of ellipsoidal particles in the  $\mu\text{g}$ -sample are shown in Fig. 11d and Fig. 11h. The corresponding FFT processed images reveal either an FCC structure (Fig. 11e–Fig. 11g) or a hexagonal structure (Fig. 11i–Fig. 11m). The pattern of the FCC structure (Fig. 11e) is oriented in the  $[011]$  zone axis with diffraction reflections close to zone axis belonging to  $\{111\}$  and  $\{200\}$  plane family. Based on this, the lattice parameter is determined to be  $5.458 \pm 0.160 \text{ \AA}$ , which is close to the lattice parameter  $a$  of hexagonal  $\text{Al}_3\text{Zr}_4$ . However, none of the so far reported cubic phases fits to this lattice parameter. Concerning the chemical composition of the phase, no significant difference compared to the spherical cubic  $\text{Cu}_2\text{Zr}_4\text{O}$  phase has been observed.

The crystalline structure (shown in Fig. 11h) could be related to a hexagonal structure oriented in  $[2\bar{1}10]$  zone axis as visible in Fig. 11i–Fig. 11m, based on the perfectly matching angle and distance relationship of visible diffraction reflections. In this case, the measured d-spacing of the  $(0002)$  reflection (Fig. 11j) is about  $2.696 \pm 0.040 \text{ \AA}$ , which results in a lattice parameter of  $5.393 \pm 0.081 \text{ \AA}$ . This value is in good agreement with the lattice parameter  $c$  of  $\text{Al}_3\text{Zr}_4$  reported in the literature [50]. This is also valid for the  $(0001)$  reflection that appears due to double diffraction.

However, other d-spacings measured from reflections at  $(01\bar{1}0)$ ,  $(01\bar{1}1)$  or  $(01\bar{1}2)$  positions and the calculated lattice parameters of  $3.446 \pm 0.015 \text{ \AA}$  do not correspond to the lattice parameter  $a$  of the hexagonal  $\text{Al}_3\text{Zr}_4$  phase [50]. In view of these results, both structures observed in the ellipsoidal particles could correspond to transition states between  $\text{Cu}_2\text{Zr}_4\text{O}$  and  $(\text{Al,Cu})_3\text{Zr}_4$  due to their lattice parameter  $5.458 \pm 0.160 \text{ \AA}$  being partially close to the lattice parameters  $a$  and  $c$  reported for  $\text{Al}_3\text{Zr}_4$ , and their chemical composition being comparable to  $\text{Cu}_2\text{Zr}_4\text{O}$  [46, 50, 51]. However, the possibility of a so far unknown phase cannot be completely ruled out.

The EDS element maps of the nanocrystalline phases and the surrounding nearby amorphous matrix for the



**Fig. 12** EDS maps of crystalline phases in the interface region of MAPHEUS-10  $\mu\text{g}$ -sample (a.-f.) and lab equivalent (g.-l.) of AMLOY-Zr01. EDS maps of  $\mu\text{g}$ -sample refer to the marked rectangular area in a. and scale bar represents 100 nm. Capital letters refer to local chemical composition given in Table 5



**Table 5** Chemical composition of nanocrystalline phases and amorphous region next to it obtained from EDS spectra of the  $\mu\text{g}$ -sample with respect to location A–E in Fig. 12a in at.%, as well as chemical composition of nanocrystalline phases obtained from EDS spectra of lab equivalent with respect to location A–E in Fig. 12g in at. %

flight sample					
Element	A	B	C	D	E
O	13.97	13.64	14.02	18.84	10.29
Al	12.39	10.05	12.81	2.65	4.45
Cu	33.36	33.15	33.32	18.63	33.07
Zr	39.97	39.98	40.18	59.31	50.94
Nb	0.99	1.19	0.68	0.57	1.26
lab equivalent					
Element	A	B	C	D	E
O	18.46	18.27	17.78	16.43	10.41
Al	13.09	15.64	13.18	15.62	9.76
Cu	31.18	32.05	31.10	32.86	29.27
Zr	36.79	33.88	37.55	34.89	49.22
Nb	0.49	0.16	0.40	0.21	1.34

$\mu\text{g}$ -sample and the lab equivalent are shown in Fig. 12. The corresponding EDS results of the highlighted crystalline and amorphous regions are presented in Table 5.

With respect to the nominal chemical composition of AMLOY-Zr01, nanocrystalline phases are depleted in Zr, but are enriched in Al and slightly in Cu, which agrees with the previous reports on the cubic phase  $\text{Cu}_2\text{Zr}_4\text{O}$  [34, 46]. In the  $\mu\text{g}$ -sample, the amorphous matrix in the close vicinity of nanocrystalline phases is either slightly depleted in Al and Zr or depleted in Cu (Table 5 D, E) compared to the chemical composition of AMLOY-Zr01. An Al and Zr depletion of the amorphous matrix is also measured in the lab equivalent (Table 5 E). In general, the chemical composition determined by EDS is similar to what de Olivera et al. [46] have measured for  $\text{Cu}_2\text{Zr}_4\text{O}$  and its surrounding amorphous matrix. Slightly more oxygen has been detected in the crystalline phases of the lab equivalent compared to the  $\mu\text{g}$ -sample. However, a nominal difference in oxygen content between the experiments cannot be fully disregarded. The Nb content is in general much lower than expected from the nominal chemical composition of AMLOY-Zr01. With regards to the phases present as a function of gravity, no difference is observed on the basis of the TEM analysis.

The characterization of the nanocrystalline phases by their stoichiometric compositions using TEM-EDS was not possible due to their size and signal overlap with the surrounding bulk matrix. Furthermore, it is possible that the TEM analysis does not identify all existing phases, since only small sections can be analyzed. Nevertheless, the EDS results revealed significant Al incorporation in the  $\text{Cu}_2\text{Zr}_4\text{O}$  phase, which is believed to be responsible for the increased

lattice constant derived from HRTEM. Moreover, the identification of the  $\text{Cu}_2\text{Zr}_4\text{O}$  phase by both TEM and SEM, and in particular the presence of a transition state between  $\text{Cu}_2\text{Zr}_4\text{O}$  and  $(\text{Al,Cu})_3\text{Zr}_4$ , indicates that the formation of the both phases could be related, despite the potentially different cooling rates. Especially, the observed more volumetric distribution of  $\text{Al}_3\text{Zr}_4$  in the bulk sample by XRD-CT may be a result of triggered formation of the  $\text{Al}_3\text{Zr}_4$  phase by the nucleation of  $\text{Cu}_2\text{Zr}_4\text{O}$  in the melt, either due to heterogeneous nucleation or due to a remaining liquid depleted in  $\text{CuZr}_2$ .

## 4 Conclusion and outlook

To summarize, using the MARS-M device, samples of AMLOY-Zr01 metallic glass compositions were built in the lab and in microgravity. Extensive analysis of both gravity states revealed complex crystalline phases among an amorphous matrix for all samples, whose fraction depends on the process parameters, in this case laser power and scanning speed used. The minimum laser powers necessary to produce samples without large pores are similar to that used in the conventional LPBF processes [27]. Also, the trend that the crystalline fraction increases with increasing laser power agrees with the observation in LPBF-processed of BMGs without a gas flow. Therefore, the technique is considered to be suited for both ground and space applications.

No significant difference – beyond the physical outer aspect – is observed between the part built in microgravity and that built in the lab. Thus, it is concluded that using the gas flow-assisted mechanism for stabilizing the powder bed, similar process conditions as the conventional LPBF can be established. Thus, the current knowledge on optimizing the process parameters for AM of BMGs can be transferred. Differences between samples processed in microgravity and on ground are mainly originated from the wetting behavior of the melt. This is an interplay between surface tension and hydrostatic pressure generated by the gas flow, which remains to be further explored for longer microgravity time, and compared with other AM approaches. One might expect convection to make a difference, as it is a parameter shown to affect the phase formation from the melt [52, 53]. However, as the cooling during LPBF is generally fast, and the viscosity of the glass forming alloys is relatively high [22], convection within the melt could be limited, whether in the lab or in microgravity.

This crystalline region is mostly present in the interlayer regions, as shown in SEM and in X-ray diffraction and tomography. Moreover, while the crystallization at the interface is known to associate with HAZ and therefore scanning strategy, the observed  $\text{Al}_3\text{Zr}_4$  phase appears to show more volumetric distribution. It has been found that the formation



of  $\text{Al}_3\text{Zr}_4$  appears to be closely related to the nucleation of the  $\text{Cu}_2\text{Zr}_4\text{O}$  phase in the melt, and hence the oxygen content in the processes. Thus, in terms of manufacturing fully amorphous parts in space, higher scanning speeds and a reduced oxygen content are necessary. The current limit of the laser scanning speed can be overcome, e.g., for long-term missions, where the acceleration of the launch is lower or sensitive laser systems can be assembled after launch, while from the mechanical strength point of view, residual oxygen content and some crystalline fraction could provide even certain advantages. In particular, for AMLOY-Zr01, it has been shown that additively manufactured sample with higher oxygen content exhibits higher contact stiffness (more resistance to shear, also higher elastic modulus) but lower fracture toughness (more brittle) [54].

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**Author Contributions** CN and JT designed and built the setup. CN, JT, MC, and JW ran measurements in the lab and during MAPHEUS campaigns. CN, JT, MC, and FY oversaw and guided measurements during the synchrotron beamtime. OG and ACD guided and performed the measurements during the synchrotron beamtime. OG processed the XRD-CT data including the reconstruction. LR and RB initiated and conducted the differential calorimetry measurements. MC and MK performed together the SEM studies and analyzed the results. CKA and CA conducted TEM investigations, analyzed, and interpreted the TEM results. MC and FY analyzed and interpreted the data. MC, FY, CN, and CA wrote the manuscript. DZ initiated and supported the TEM investigations. AM initiated and supported the project as well as procured the flight opportunities. All authors revised the manuscript.

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**Data Availability** The datasets used and/or analyzed during the current study are available from the corresponding author on reasonable request.

## Declarations

**Conflict of interest** The authors declare no conflict of interest.

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