

1 *In-situ* hot forging direct energy deposition-arc of CuAl8 alloy

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13 **Abstract**

14 CuAl8 alloy finds applications in industrial components, where a good anti-corrosion
15 and anti-wearing properties are required. The alloy has a medium strength and a good
16 toughness with an elongation to fracture at room temperature of about 40%.
17 Additionally, it has a good electrical conductivity, though lower than that of pure Al or
18 pure Cu. Despite these characteristics, additive manufacturing of the CuAl8 alloy was
19 not yet reported. In this work, the direct energy deposition-arc (DED-arc) with and
20 without *in-situ* hot forging was used to determine the microstructure evolution and
21 mechanical properties. No internal defects were seen on the parts produced. Hot
22 forging combined with DED-arc was seen to reduce and homogenize the grain size,
23 improve mechanical strength and isotropy of mechanical properties. Moreover, the use
24 of this novel DED-arc variant was seen to reduce the magnitude of residual stresses
25 throughout the fabricated part. We highlight that this alloy can be processed by DED-
26 arc, and the hot forging operation concomitant with the material deposition has
27 beneficial effects on the microstructure refinement and homogenization.

28 **Keywords:** CuAl8 alloy; directed energy deposition-arc; forging; viscoplastic
29 deformation; grain refining.

1^{1#} One of the authors of this article is part of the Editorial Board of the journal. To avoid potential
2 conflicts of interest, the responsibility for the editorial and peer-review process of this article lies with
3 the journal's other editors. Furthermore, the authors of this article were removed from the peer review
4 process and had no, and will not have any access to confidential information related to the editorial
5 process of this article.

* AM - Additive Manufacturing; DED-arc - Direct Energy Deposition-arc; DAC - Distance to Arc Centre;
FCC - Face-Centred Cubic; GMAW - Gas Metal Arc Welding; HF-WAAM - Hot Forging WAAM;
NDT - Non-Destructive Testing; WAAM - Wire and Arc Additive Manufacturing

301. Introduction

31Additive Manufacturing (AM) is being intensively investigated by several research
32groups due to the intrinsic characteristics technological processes that are in line with
33industrial digitalization, reduction of energy and raw materials consumption and clean
34forms to manufacture goods [\[1,2\]](#). Directed Energy Deposition process category as
35defined by international standards, is also commonly known as wire and arc additive
36manufacturing (WAAM). From the wide range of AM processes for metallic alloys,
37WAAM is eventually the least expensive as it uses an electric arc as the heat source.
38Moreover, the equipment and the well-consolidated knowledge from arc welding
39technology and welding metallurgy are readily applied in WAAM [\[3\]](#).

40In multiple pass arc welding of metallic materials without solid state transformations,
41the development of coarse-grained structures by epitaxy is well known. The same is
42observed in multilayer WAAM process. Solidification structures, either cellular dendritic
43or columnar ones, develop through the successive layers and this is typically
44accompanied by a detrimental effect on the mechanical properties and performance of
45the fabricated components [\[4\]](#).

46Several groups have developed variants based on plastic deformation and active
47cooling to improve the as-deposited microstructure [\[5–8\]](#). Regarding the plastic
48deformation variants they can be categorized accordingly to the temperature at which
49they are performed (hot or cold) and considering how they are applied, i.e. continuously
50or discontinuously.

51The developments in continuously hot deformation during WAAM usually comprises a
52roller that follows the torch at a fixed distance and deforms the material at high
53temperatures. Zhang et al. [\[9\]](#) studied the effects of the roller-heat source distance on
54the samples surface aspect and verified that for a short distance, the deformation
55temperature is too high and the material adhere to the roller and instead of being
56deformed it is dragged by the roller, resulting in a bad final surface aspect. In contrast,
57at a higher distance the deformation is insufficient due to the high colling rate and the
58lower rolling force. Therefore, a fine tuning of the roller-heat source distance must be
59performed. Despite the difficulties in tunning the deformation parameters several
60authors have reported that continuously hot deformation variant is able to refine the
61microsctructure, Nickel-based superalloy 718 [\[10\]](#) and the Ti-6Al-4V alloy [\[11\]](#).

62However, the continuously hot deformation using a roller have limited applications,
63since the rolling system cannot rapidly change the direction, it limits its positioning,
64increases the complexity of the deposition path, or even makes it impossible to
65produce some complex geometries.

66With regard to the *in-situ* discontinuous hot forming variants, most authors used
67systems similar to the hammer peening technique, where a an electromagnetic
68actuator [12], or a pneumatic cylinder [13], are coupled to a steel impact tool used to
69deform the bead surface. Ye et al. [14] applied a discontinues hot forming during the
70production of Ti-6Al-4V alloy, and verified that it causes a microstructural
71recrystallization forming an equiaxed structure with reduced the grain size, and
72improves the microhardness of top surface. However, it was verified that the developed
73system was promoting the formation of defects such as pores and cracks, and have
74only studied the effects of the deformation on single bead, the effect of the following
75deposition on the recrystalized microstructure was not studied. With the same
76mechanism Li et al. [15] studied the effects on microstructure of single-bead GH3039
77superalloy, achieving also a grain refinement and an increase of 46 % in the material
78microhardness, attributed to the dynamic recrystallization induced by the plastic
79deformation.

80Duarte et al. [13] developed a variant based on *in-situ* hot forging in the viscoplastic
81regime with considerably lower loads than the used in the variants presented above,
82and described the fundamentals of the process. It encompasses a linear hammer, with
83a stroke of about 10 mm and operating at 5 - 10 Hz, that performs a locally viscoplastic
84deformation of the already deposited layers immediately after the material is deposited.
85By plastically deforming the as-deposited layer at high temperature, dynamic
86recrystallization of the previous layer is promoted. Moreover, process-related defects
87such as pores, can be eliminated, and the resulting flat surface facilitates the
88application of Non-Destructive Testing (NDT) [16]. Therefore, by applying deformation
89while the material is at high temperatures (in its viscoplastic regime) the
90aforementioned features can be achieved with a fraction of the load that would be
91required if the deformation was performed at low temperatures, i.e., near room
92temperature.

93The presence of recrystallized grains in WAAM manufactured parts has several
94advantages: first, it can increase the mechanical strength following the Hall-Petch
95relationship; then, the existence of refined grains at the top of the deposited layer
96provides a higher density of nucleation sites, thereby decreasing the susceptibility to

97large grain growth as typically observed in WAAM deposits. This is particularly relevant
98for alloys without solid state transformations, which are more susceptible to significant
99grain growth during successive thermal cycles [3].

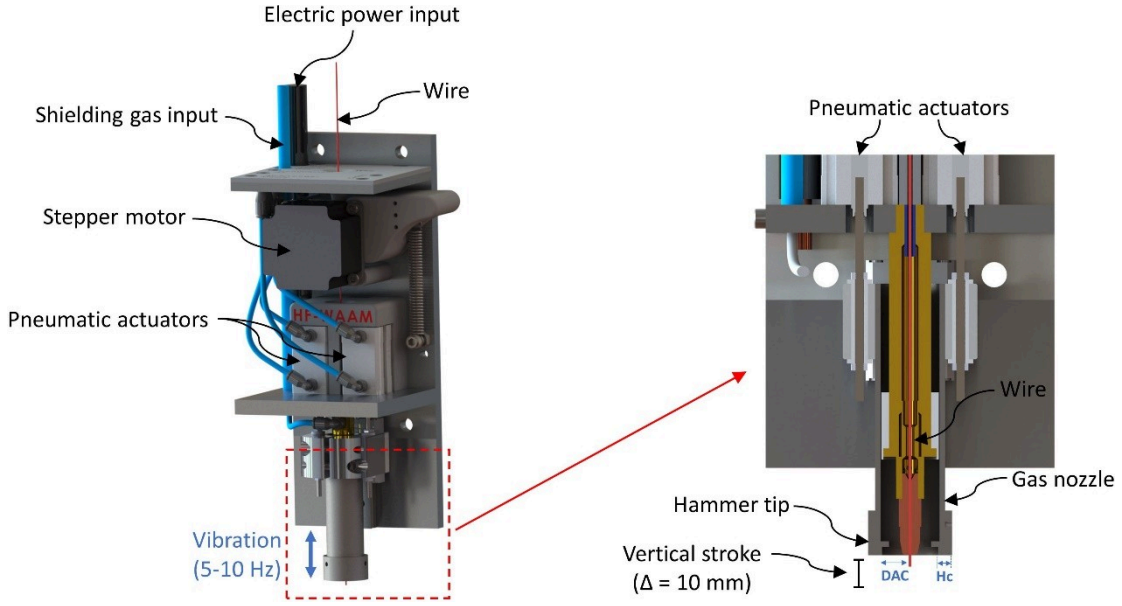
100Linear *in-situ* discontinuous hot forming variants were already designed and tested with
101very promising results, but the linearity of the systems has two major shortcomings: it
102imposes a limitation of the part geometries to linear ones, and the contact area
103between the hammer and the material surface forged was restricted. Thus, in this work,
104a new hammer was designed as a circular crown placed concentrically with the welding
105nozzle (refer to Figure 1) allowing to forge nonlinear deposition paths since the forging
106action is independent of the travel direction of the torch. The developed equipment
107characterized and the impact of *in-situ* Hot Forging WAAM (HF-WAAM) on the
108microstructure and mechanical properties was probed by microscopy, high energy
109synchrotron X-ray diffraction, electric conductivity, and mechanical testing.

1102. *In-situ* Hot Forging WAAM torch description

111The technological aspects of the innovative WAAM variant (patented by the authors of
112this research [17]) comprises a pneumatic system that actuates the hammer tip and
113forges the material immediately after its deposition. The hammer (detailed in Figure 1) is
114actuated by two pneumatic actuators placed symmetrically to avoid discrepancies in
115the generated forces. The pneumatic system consists of: two Festo ADN-12-10-I-P-A
116pneumatic cylinders; one Festo MS-LFR air filter; one pressure regulator and gauge;
117and one 5/2-way bi-stable solenoid valve Festo VUVS-LK20. Hot forging during
118deposition requires the control of a set of process parameters that may influence the
119material condition, geometry, as well as metallurgic and mechanical properties. These
120include forging force, F_F [N]; forging frequency, F_f [Hz]; distance to arc centre DAC
121[mm], and hammer geometry. Concerning the DAC, it is necessarily defined by the
122radius of the hammer tip (Figure 2b).

123The area forged in each hammer stroke depends on the bottom surface geometry of
124the hammer and the forging step, F_s [mm], (i.e., the distance, in millimeters, travelled by
125the hammer in one cycle); which is a function of the F_f and the travel speed, TS [mm/s].
126Thus, the forging step is given by Eq. 1.

$$F_s = \frac{1}{F_f} \times TS \quad \text{Eq. 1}$$



129Figure 1 - Schematic representation of the developed torch for in-situ Hot Forging WAAM (HF-WAAM). A
130video of the system can be seen in supplementary material section (temporarily here: [Folder](#)).

131Since the hammer has a circular crown shape, the forging step must be lower than the
132crown thickness to avoid unforged areas between consecutive steps. To determine the
133applied forging pressure, it is necessary to identify the forged area during each cycle.
134For this particular geometry, the forged area is represented in [Figure 2](#) and can be
135easily estimated by [Eq. 2](#), were F_s [mm] is the above-mentioned forging step ([Eq. 1](#))
136and L_w [mm] is the arc length defined by the contact between the hammer circular
137crown and the deposited material. The L_w [mm] can be computed by Equation 3, were
138 R_m [mm] is the hammer internal radius and β [°] is the angle of the arc half-length. β [°]
139can be computed by [Eq. 4](#), where W [mm] is the bead width.

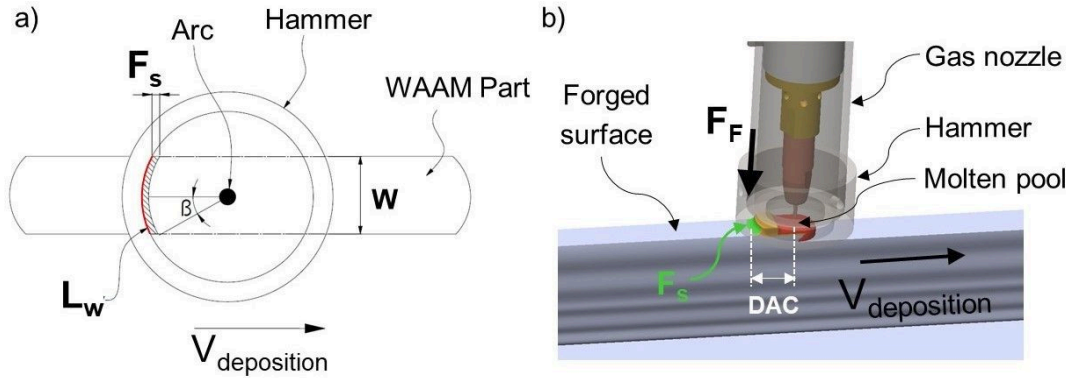
$$A = F_s \times L_w \quad \text{Eq. 2}$$

$$L_w = \frac{2\pi R_m \times 2\beta}{360} \quad \text{Eq. 3}$$

$$\beta = \sin^{-1} \left(\frac{W}{2R_m} \right) \quad \text{Eq. 4}$$

140Another important feature of the hammer is the dimension of the circular hammer
141crown, H_c [mm], compared to the forging step, F_s [mm]. In fact, H_c must be greater
142than F_s to assure that the previously deformed material will act as a mechanical
143stopper, avoiding excessive deformation of the viscoelastic deposited material ([Figure](#)
144[2](#)). This promotes a continuous flat surface, that is highly dependent on the material

145temperature. Slight deviations of this temperature promote different forging depths, and
 146therefore, an uneven upper surface. Three videos of the *In-situ* Hot Forging WAAM
 147torch can be seen in supplementary material section (temporarily here: [Folder](#))



148

149Figure 2 - Schematic representation of the forged area at each step: a) top view (2D), b) isometric view
 150(3D). The forged projected area is computed by Equation 2.

151

152 3. Dynamic characterization of the forging system

153The speed of the hammer when it touches the deposited material, is an important
 154variable of the forging process, as it determines the kinetic energy carried by the
 155hammer, that is later absorbed by the as-deposited layer under the form of
 156deformation. The speed of the hammer also determines the forging force. The
 157theoretical final speed of the hammer, v_f [m/s], can be calculated using the equations of
 158the uniformly accelerated motion ([Eq. 5](#)), where: a [m/s²] is the hammer acceleration
 159and ΔY [m] is the vertical distance traveled by the hammer, correspondent to the stroke
 160of the pneumatic cylinders.

$$v_f = \sqrt{2 a \Delta Y} \quad \text{Eq. 5}$$

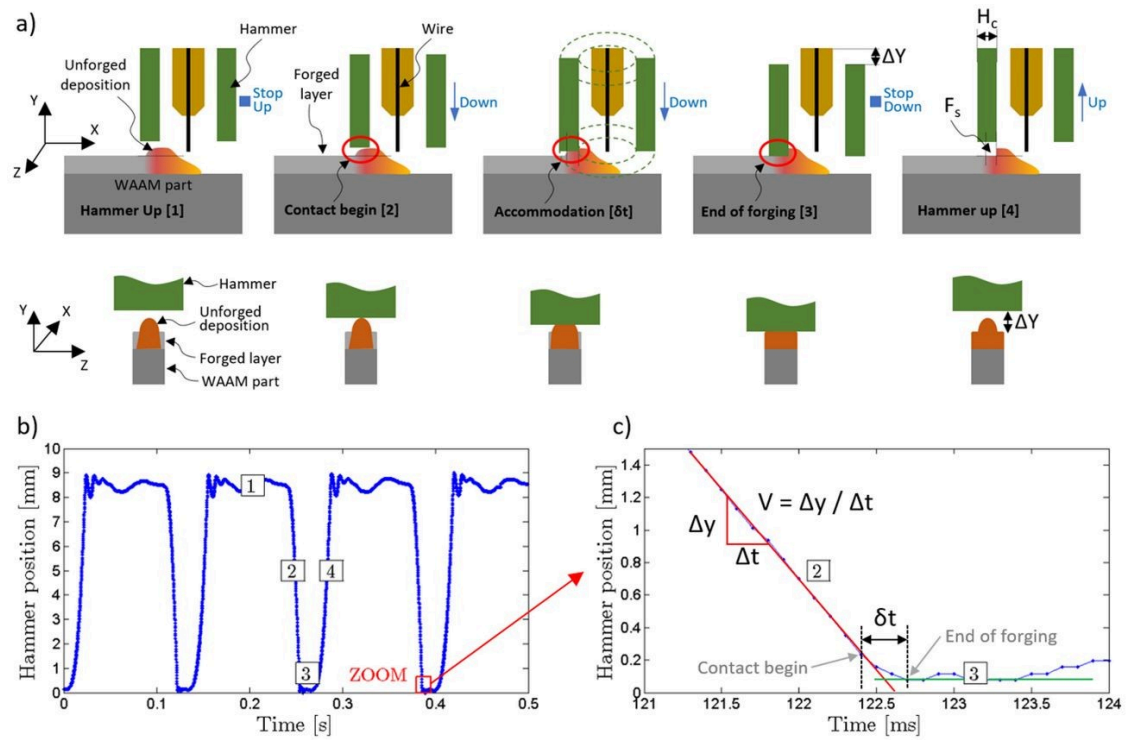
161However, the hammer's acceleration is difficult to compute due to the unknown friction
 162in the moving parts and the unknown real dynamic pressure inside the pneumatic
 163cylinders. Therefore, an alternative approach to measure the final speed of the hammer
 164is needed. The velocity was experimentally measured, and the dynamic
 165characterization of the forging mechanism was performed by high-speed imaging. The
 166acquisition of the images was obtained during the vertical movement of the forging
 167system, when a sample was being produced. This method was chosen over others, as
 168it allows the simultaneous evaluation of several factors, such as: the material strain

rate, the estimation of the impact force, and the evaluation of the overall performance of the forging system.

A video of the movement of the hammer was acquired using a high-speed camera (Photron Mini WX50) operating at 10,000 fps and 512 x 256 pixels, thus having a temporal resolution of 0.1 ms and a spatial resolution of 0.0382 mm [Figure 3 a](#)) depicts the schematic representation of the movement of the hammer during forging. It operates according to 4 distinct stages (also refer to [Figure 3 b](#)): 1-hammer is stopped at the top position; 2-hammer is descending with increasing velocity; 3-hammer is forging; and 4-hammer is moving up. It is worth to notice that a duty cycle of 20% is used to reduce the time contact between the hammer and the hot deposited material.

From the acquired data ([Figure 3 c](#)), it was observed that the impact contact between the hammer and the part occurs almost instantaneously during a very short period (δt). The material deformation starts when the hammer comes into contact with the as-deposited layer. Although the hammer has acceleration through its descent movement, the instantaneous velocity before impact can be computed by $V = \Delta y / \Delta t$, since at this stage of the movement the velocity of the hammer is almost constant ([Figure 3 c](#)). During this initial deformation, the deposited material loses its natural convex shape and accommodates to the planar surface of the hammer, increasing the contact area and therefore the resistance the material imposes to the hammer movement ([Figure 3 a](#)). With the increase of this resistance during δt , the hammer starts to decelerate until it comes into contact with the previously deformed material which is at a lower temperature and acts as a stop for the hammer (stage 3).

The last stage of the deformation is critical in the process, since it is at this stage of deformation, when the hammer decelerates, and the hammer kinetic energy is transferred to the part in production. It must be noticed that the forging force is not a static force, resulting from the static pressure provided by the pneumatic actuators; it is rather an impact force, resulting from the almost instantaneous impact of the mass of the hammer into the deposited material. Consequently, the forging force is considerably higher than the actuating force (pressure \times area) of the pneumatic actuators.



199

200 Figure 3- Dynamics of the in-situ HF-WAAM during a deposition with forging at $F_f = 8$ Hz and a duty cycle
 201 of 20%. a) Schematic representation of the movement of the hammer during forging, b) Hammer position
 202 along the time during the deposition and forging of a sample acquired at 10,000 fps, c) Zoom of b). A video
 203 of the forging deposition can be seen in supplementary material section (temporarily here: [Folder](#)).

204 Thus, the impact forging force depends on: the hammer speed before contact ($V = \Delta y /$
 205 Δt) and the impact time, δt , both represented in [Figure 3 c](#)). It can be estimated by
 206 using the principle of impulse and momentum ([Eq. 6](#)), where I [N·s] is the impulse,
 207 P [N·s] is the linear momentum, m [kg] is the mass of the moving hammer parts, Δv [m/
 208 s] is the difference between the initial and final velocity of the forging hammer, F_{avg} [N]
 209 is the average force during the impact (forging), and δt [s] is the time duration of the
 210 impact.

$$I = \Delta P = m \cdot \Delta v = F_{avg} \cdot \delta t \quad \text{Eq. 6}$$

211 In the present case, the mass of the moving parts of the torch is 0.435 kg, the average
 212 experimental impact time, δt is 0.475 ms, and the final velocity of the hammer is
 213 1.12 m/s. Therefore, the average impact force is of 1026 N. The forging strain rate is
 214 22.2 s^{-1} , considering a plane strain compression in the plane ZOY with the initial and
 215 the final layer height of 1.45 and 1.27 mm, respectively.

216 Considering an average measured width of the hot forged sample of 7.3 ± 0.3 mm and
 217 that the forging step is of 0.75 mm (i.e. in each stroke the hammer advances 0.75 mm),
 218 the area that is forged in each hammer stroke is 5.475 mm^2 . Thus, the stress exerted
 219 by the hammer in the deposited material is of about 190 MPa.

Figure 4 details typical flow-stress curves for this alloy obtained at a strain rate of 5 s^{-1} , as well as the values of the stress and strain achieved during the *in-situ* hot forging. Despite the difficulty in obtaining the flow stress curves of this material at strain rates near those experienced by the material, the following qualitative conclusions are valid, as the increase of the strain rate increases the material flow value stress.

An equivalent true strain of 0.17 was achieved during the *in-situ* hot forging which was calculated using the measurements of the layer height of the hot-forged and as-built samples. By superimposing the calculated true strain and the stress exerted by the hammer (190 MPa) on the flow stress curves of this material (refer to [Figure 4 a](#)), it is possible to verify that the hot forging must occur while the material is in the 700 to 750 °C temperature range. Additionally, the thermographic analysis shown in Figure 4 b) confirms that the material is forged at a temperature of 750 °C. It is also observed that the hammer does not exceed a temperature of 150 °C, which justifies the choice of a tool steel for the hammer material, since at this temperature the material does not lose its properties.

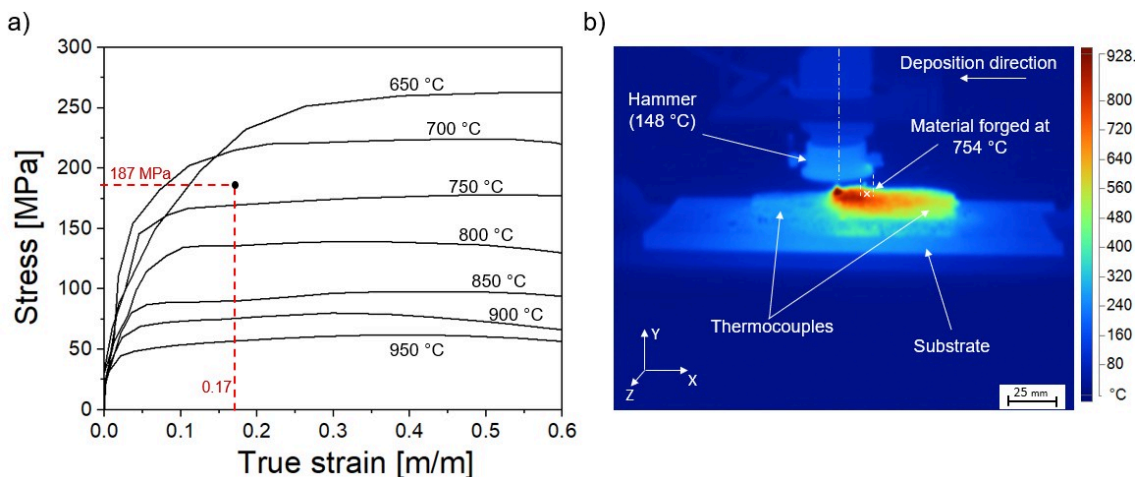


Figure 4 - Stress-strain curves at various temperatures for the CuAl alloy (a); thermogram of the deposition, with the identification of the hammer temperature and the material temperature while forged (b).

Since the melting temperature range of this alloy is between 1053 and 1100 °C, it can be concluded that it is possible to increase the deformation imposed on the material by reducing the diameter of the hammer, which implies that the material is forged while it is at a higher temperature, since the stress applied will remain constant, or through the increase of the forging frequency, which makes that in each hammer strike the forged area is smaller and therefore the applied stress increases.

245Thus, it is demonstrated that the developed *in-situ* hot forging system can be adjusted
246as needed, and plastic deformation during WAAM can be performed with the material
247at different temperatures which in turn allows to tune the amount of deformation
248imposed during the processing and control the microstructure evolution, i.e., by
249preferentially inducing strain hardening, if forging is applied at low temperature, or by
250promoting dynamic recrystallization when the deformation is at high temperature
251regimes.

254. Materials and methods

253The material used in this research was a CuAl8 alloy which is a cooper-based alloy
254with 8.5 wt. % of aluminium. It also contains a small addition of iron that promotes the
255formation of precipitates homogeneously distributed throughout the structure acting as
256grain refiners during solidification, thus improving the alloy mechanical properties [18].
257The chemical composition of the alloy is depicted in Table 1.

258Table 1 - Chemical composition of the CuAl8 wire [wt. %]

	Cu	Al	Mn	Ni	Fe
CuAl8	Bal.	8.5	<1.0	<1.0	1

259The setup used to produce the samples consisted of a customized DED-Arc torch
260attached to a moving head within a working envelope of 2760 × 1960 × 2000 mm³. A
261PRO MIG 3200 power source from KEMPY was used to deposit 1 mm diameter wire of
262CuAl8 on a mild steel substrate.

263For the dynamic characterization, a Photron FASTCAM Mini WX50 high-speed camera
264equipped with a Nikon AF IKKOR 28-105 mm macro lens was used. To ensure the
265required levels of lighting, four 100 W LED floodlights from V-TAC were used. The data
266obtained with the high-speed camera was processed using a Python-based software,
267in which each acquired frame is binarized with a threshold function, and the position of
268the forging system is targeted.

269Two sets of samples, one as-built and the other hot forged were produced to study the
270effect of hot forging. The length of each produced sample was fixed at 100 mm and the
271time interval between layers was fixed at 15 s. The walls were built with a zig- zag
272deposition strategy, wherein the deposition of a layer started at the end point of the
273previous one.

274The process parameters were the same for both samples and are detailed in Table 2,
275the only difference being the application of the hammer forging in one of the samples.

276Table 2 – Process parameters used during the deposition

Deposition parameters	
Welding mode	GMAW – continuous mode
Number of layers	10
Wire feed speed	4 m/min
Travel speed	360 mm/min
Voltage	19 V DC +
Contact tip to work distance	9 mm
Shielding gas	Argon 99.99 %
Gas flow rate	15 l/min
Hot forging parameters	
Forging frequency	8 Hz
Forging pressure	0.5 MPa
Distance to arc center	12.5 mm

277Cross-sections from the center of each sample were cut, polished and etched. A *Leica*
 278*DMI 5000 M* inverted optical microscope was used to analyze the microstructural
 279features. The estimation of phase percentage was done using a *Python* software
 280developed in-house. Hardness tests were performed with a load of 4.9 N along the part
 281height. The distance between indentations was of 1 mm.

282Electrical conductivity was assessed by eddy current measurements to evaluate the
 283effect of the *in-situ* hot forging on this property, relevant for electromagnetic
 284applications, and provide a complementary information to hardness [19]. An Olympus
 285Nortec 600D impedance measurement equipment and an absolute helical shielded EC
 286probe, with 3 mm diameter, operating in bridge mode, were used. The test frequency
 287was set at 350 kHz, and the angle at 219 ° to align the imaginary axis with the
 288variations in electrical conductivity.

289Uniaxial compressive tests were performed on an Autograph Shimadzu machine model
 290AG500Kng equipped with a Shimadzu load cell SFL-50kN AG with a total capacity of
 29150 kN. A crosshead displacement speed of 0.01 mm/s was imposed. Three specimens
 292for each condition were evaluated. Reduced specimens for mechanical testing had to
 293be used due to the dimensions of the built part. These were removed from vertical and
 294horizontal directions of both samples.

295Thermal analysis was performed with a thermographic infrared (7.5 – 14 μm
 296wavelength) camera Fluke TI400 to monitor the temperature of the sample and

297hammer. The camera had an accuracy of ± 2 °C, a measurement limit of -20 to 2981200 °C, a refresh rate of 9 Hz, and a resolution of 320×240 pixel. The emissivity was 299set to 0.62, previously validated using thermocouples.

300X-ray diffraction was performed at beamline P07 of the High Energy Materials Science 301(HEMS), PETRA III/DESY, using a wavelength of 0.1423 Å (87.1 keV) and an incident 302beam of 1×1 mm. A Perkin-Elmer detector, with a pixel size of $200 \times 200 \mu\text{m}^2$, was 303placed at 1.40 m from the sample. LaB6 powder was used for calibration. The raw 2D 304Debye-Scherrer images provide qualitative information on the grain size and texture 305of the analyzed material [15].

306Moreover, from the X-ray diffraction tests performed, the residual stresses of both as 307built and hot forged samples were calculated. An in-house python-based routine with 308the xrdfit package [20] was used to implement a Pseudo-Voigt profile function to fit the 309diffraction peaks and extract, for each analyzed point, the position of the peak situated 310at $2\theta = 4.4^\circ$ which corresponds to the Cu phase (FCC- α), the predominant phase 311present on both samples, as observed in the diffractogram presented in Figure 9.

312In each sample, the scans were performed in the centerline of the wall cross section. 313The scans started in the interface between the substrate and the deposited material 314and finished at the top of the wall, with each point spaced by 0.5 mm from the previous 315one.

316The specific direction lattice spacing (d_k) was calculated from the peak position of each 317acquired point (k), according to the Bragg's law (Eq. 7).

$$d_k = \frac{\lambda}{2 \sin \theta_k} \quad \text{Eq. 7}$$

318where θ_k is the peak maximum position and λ is the beam wavelength (0.1423 Å).

319The accurate determination of the stress-free lattice spacing (d_0) is of great importance 320to calculate the absolute value of the measured residual stresses, however, for this 321material it has not yet be reported in the bibliography. Moreover, it is expected that the 322stress-free lattice spacings vary as a function of position [21], and to accurately 323measure it, it would be necessary to cut a series of small stress-free cubes to 324determine d_0 in each analysed point. Therefore, since the goal of this work is to 325evaluate the relative changes that hot forging may promote on the deposited material in

comparison the convention WAAM process, the stress-free lattice spacing considered for the following calculations was the average of the lattice space measured along the centre line of the as-built sample integrated along the full azimuthal angle.

From the azimuthal integration at 0 and 90° for each analyzed point, it is only possible to obtain the values of d_k^x and d_k^y , respectively. Therefore, the value of the lattice spacing in the d_k^z direction (sample thickness) can be calculated by [Eq. 8](#).

$$d_0 = \frac{1-\nu}{1+\nu} d_k^z + \frac{\nu}{1+\nu} (d_k^x + d_k^y) \quad \text{Eq. 8}$$

where the poison coefficient $\nu = 0.328$ [\[22\]](#).

Then, the principal strains (ϵ) can be computed from the distance between adjacent oriented planes of atoms [\[23\]](#), using the crystal lattice as:

$$\epsilon_{xx,k} = \frac{d_k^x - d_{0,k}}{d_{0,k}} \quad \text{Eq. 9}$$

$$\epsilon_{yy,k} = \frac{d_k^y - d_{0,k}}{d_{0,k}} \quad \text{Eq. 10}$$

$$\epsilon_{zz,k} = \frac{d_k^z - d_{0,k}}{d_{0,k}} \quad \text{Eq. 11}$$

Once the three principal strain components are computed the principal stresses (σ) on each plane can be calculated using the triaxial form of the Hooke's Law ([Eq. 12](#) to [14](#)).

$$\sigma_{xx,k} = \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu)\epsilon_{xx,k} + \nu(\epsilon_{yy,k} + \epsilon_{zz,k})] \quad \text{Eq. 12}$$

$$\sigma_{yy,k} = \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu)\epsilon_{yy,k} + \nu(\epsilon_{xx,k} + \epsilon_{zz,k})] \quad \text{Eq. 13}$$

$$\sigma_{zz,k} = \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu)\epsilon_{zz,k} + \nu(\epsilon_{xx,k} + \epsilon_{yy,k})] \quad \text{Eq. 14}$$

Finally, the equivalent von Mises stress was also calculated for each sample according to [Eq. 15](#).

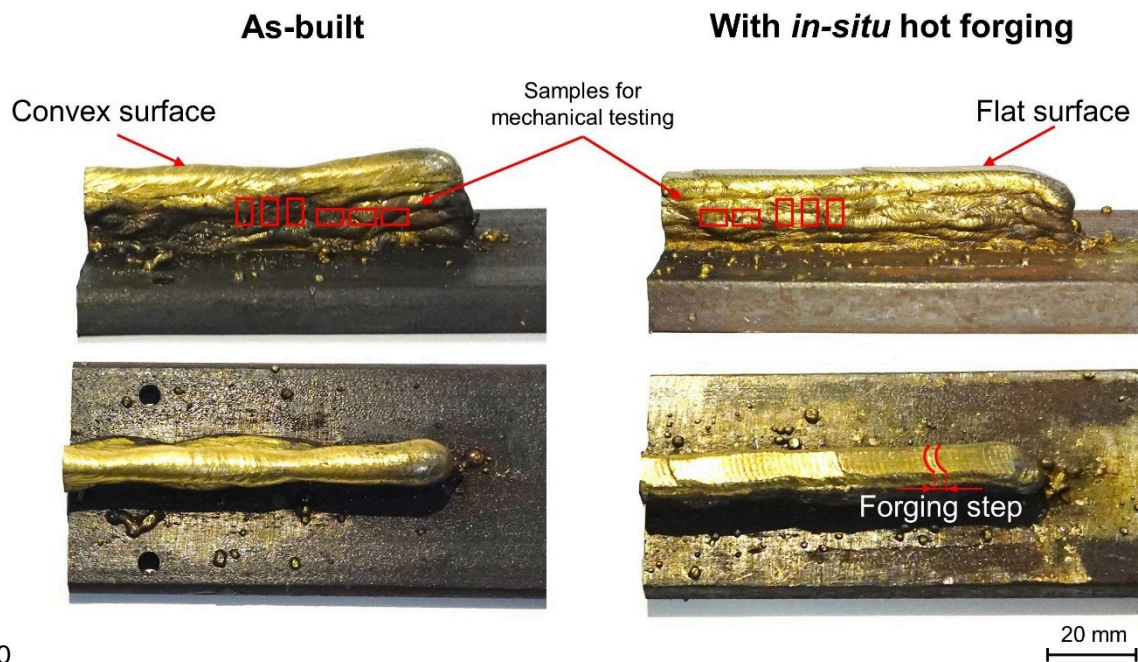
$$\sigma_{eq} = \sqrt{\frac{1}{2} [(\sigma_x - \sigma_y)^2 + (\sigma_y - \sigma_z)^2 + (\sigma_x - \sigma_z)^2]} \quad \text{Eq. 15}$$

339 **5. Results and discussion**

340 [Figure 5](#) depicts the upper and side surfaces of the produced samples. In the as-built
341 sample the difference in height at the beginning of each layer occurs due to rapid
342 cooling and the small amount of heat accumulated, in contrast to the remaining of the
343 sample where the colling conditions are stable and constant. Usually, to correct this, it
344 is required an in-line parametric correction. It is evident that the hammer has a
345 considerable impact on the sample geometry as it flattens the layers, thereby
346 diminishing this feature that needs correction. In the hot forged sample, the flat top
347 surfaces are achieved due to the constant process conditions of the *in-situ* hot forging.

348 The temperature at which the material is forged is controlled by the following process
349 conditions: deposition parameters, that define the temperature gradient along the
350 horizontal direction; and the distance from the hammer to the electric arc. Since both
351 conditions are kept constant, the material forging temperature and, therefore, its
352 properties are constant, promoting an evenly distributed flat surface. Moreover, this
353 flattening effect of the deposited layer assists in the deposition of the subsequent layer.

354 The geometry of the forged sample may change depending on various parameter. If
355 the deposition parameters are set with a higher heat input, the forging temperature is
356 higher and more deformation will be imposed, increasing the layer width and reducing
357 its heigh. Additionally, to increase the deformation, the hammer can also be placed
358 closer to the electric arc and the pneumatic pressure can also be increased.



360

361 Figure 5 – Photograph of the as-built and *in-situ* hot forged samples showing the upper and side surfaces
 362 of the as-built walls

363 The alloy did not present significant problems under WAAM, with the internal structure
 364 being free of porosities and cracks as expected [24]. However, the high thermal
 365 conductivity of these alloys tends to produce irregular lateral surfaces, and a part built
 366 in height tends to be irregular due to non-confined material flow, as shown in the
 367 macrographs of Figure 6.

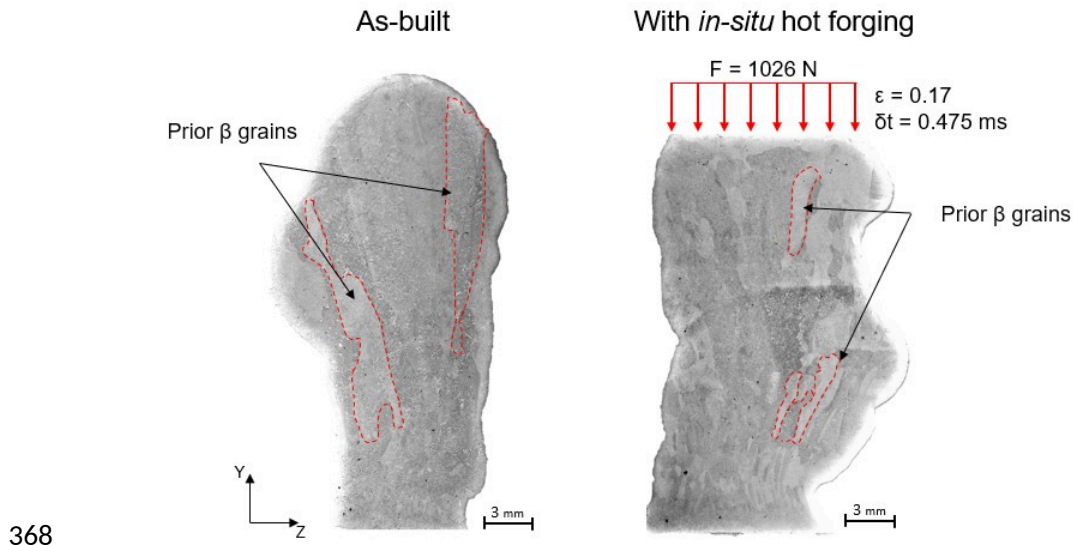


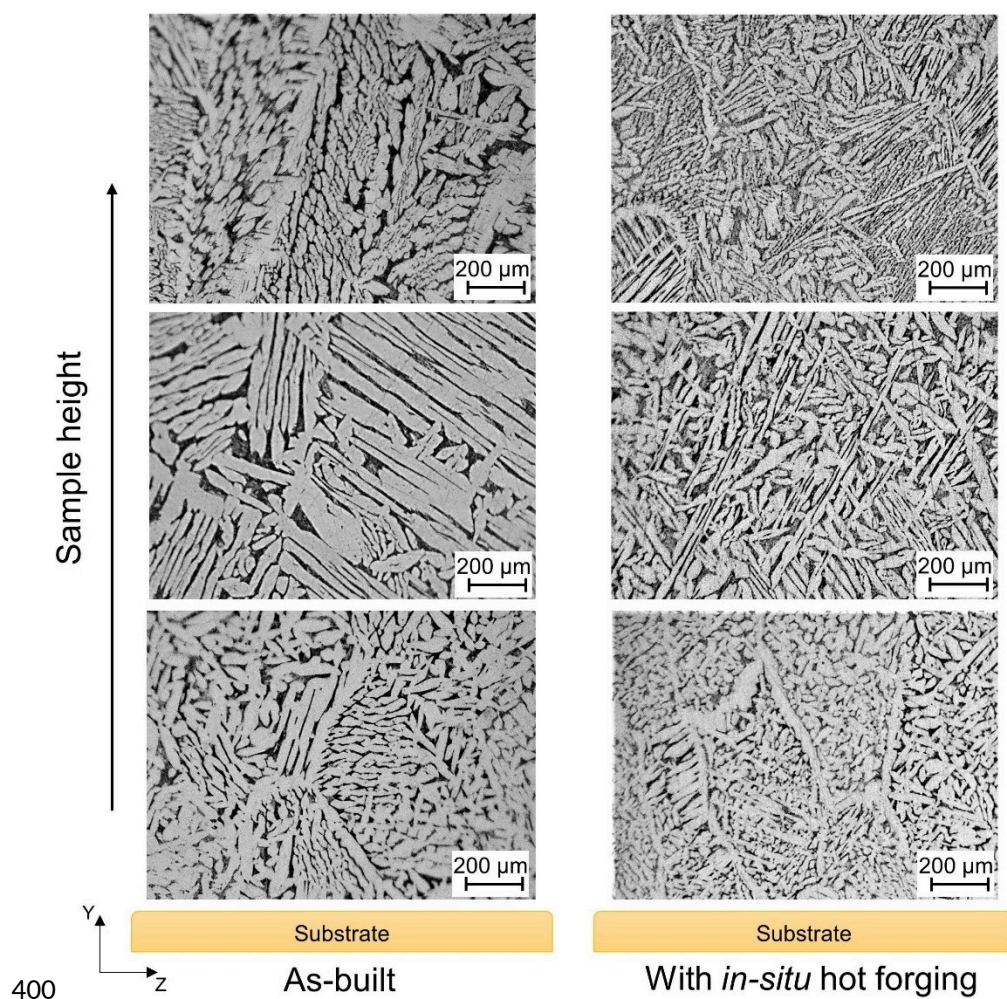
Figure 6 - Macrographs of parts built without and with hot forging showing the geometrical features.

Microstructure

During additive manufacturing of metals, the deposition of a layer causes repeated thermal cycles in the previously deposited layers which can significantly affect their microstructure, promoting different grain sizes along the build direction [25,26]. At the bottom of both samples a finer microstructure was observed, which is caused by the rapid cooling due to the heat dissipated to the cold substrate. The middle layers have the coarser grains of the sample, because the heat flow is lower than in the first layers, and therefore the cooling rate is also lower, in addition to the thermal cycles caused by the following depositions that promote grain growth. Lastly, in the upper layers, the grain size is smaller than in the middle layers, since they are subjected to less thermal cycles than the layers below.

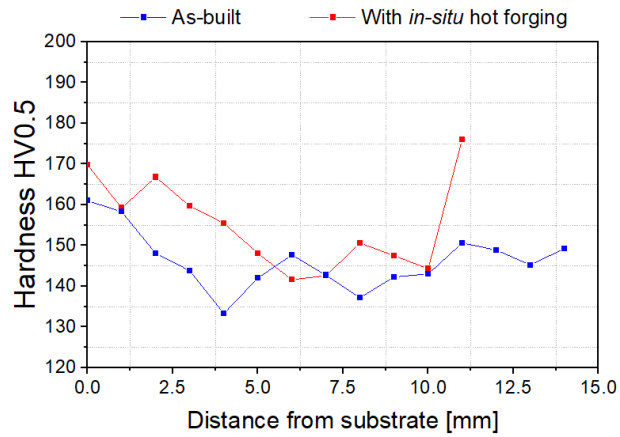
However, looking into the microstructures depicted for both samples in Figure 7, it is evident that the hot forged AM sample has a finer microstructure and that it is uniform in height, e.g., it is almost independent of the number of layers deposited on top of each other, while the as-built part has a coarser microstructure. Additionally, both samples have a primary β phase (dark regions) in the grain boundaries and α phase (bright regions) inside the grains in a cellular-dendritic morphology. The α phase has an FCC crystal structure while β phase has a BCC structure [27]. The identification of the alpha and beta phases were made according to the existing literature [28,29].

390 The average columnar dendrites width measured on the as-built sample was of
 391 $123.2 \pm 5.2 \mu\text{m}$, while in the hot forged sample was of $11.4 \pm 2.3 \mu\text{m}$, corresponding to a
 392 grain size reduction of $\approx 51\%$. Another interesting feature is that the forged sample has
 393 a higher β phase fraction, which can be seen in the micrographs ([Figure 7](#)) but also in
 394 the diffractograms depicted in Figure 9. Previous studies on Cu-Al alloys [\[30\]](#), showed
 395 that pressure may affect the solid state transformation and consequently the phase
 396 fraction, i.e. with an increase in pressure there is an increase in β phase volume
 397 fraction and vice versa. The in-situ hot forging promotes a momentary increase of the
 398 material pressure while it is at 750°C , which affect the phase transformation and
 399 increases the volume fraction of the β phase.



401 Figure 7 - Microstructure of samples produced: as built and with hot forging. α phase (white) and β (dark).

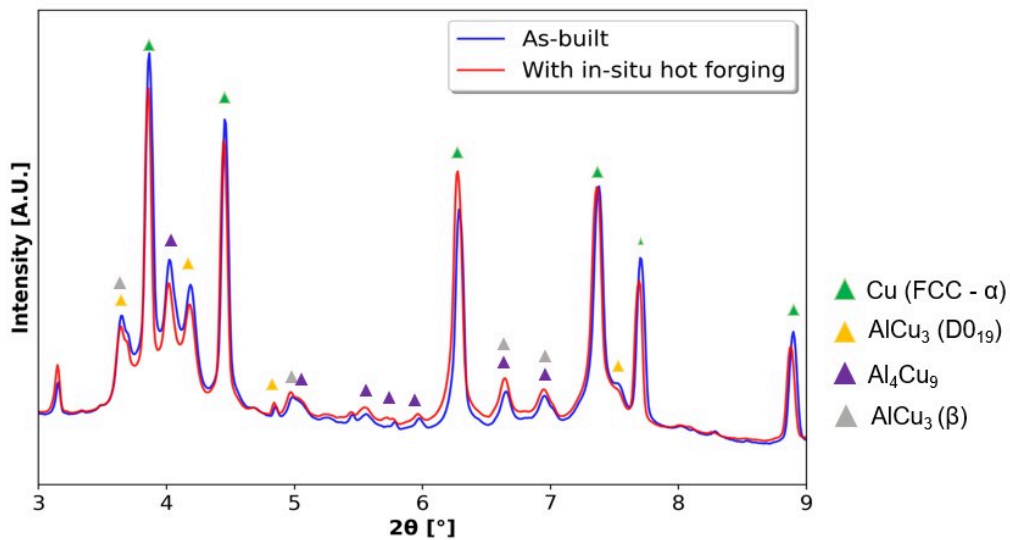
402 Hardness values are of $145 \pm 6 \text{ HV}$ and $155 \pm 9 \text{ HV}$ for the as-built and the *in-situ* hot
 403 forged samples respectively, as shown in [Figure 8](#). The increase of β phase fraction
 404 may explain the slight increase in hardness despite the significant reduction of grain
 405 size.



406

407Figure 8 - Hardness measurements across samples height.

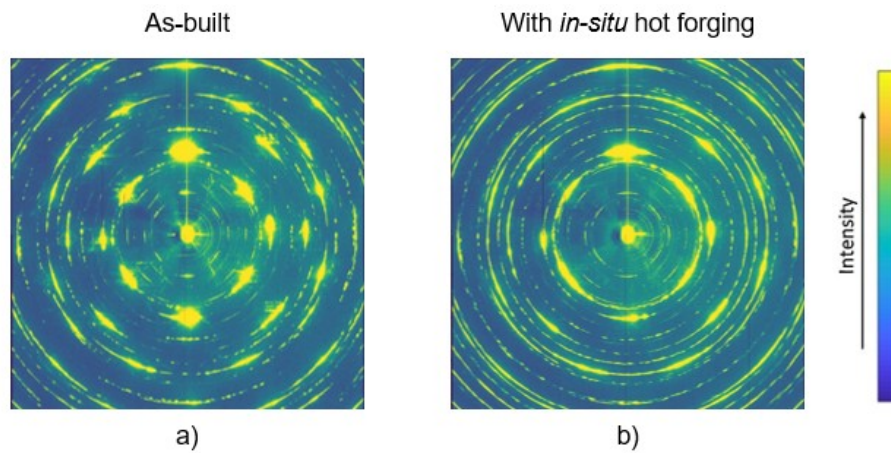
408X-ray synchrotron radiation also shows that the *in-situ* hot forging operation has no
 409effect on the identified phases as shown in [Figure 9](#). However, from the 2D Debbye-
 410Sherrer diffractograms it can be observed that the as-built part has a coarser and
 411textured microstructure, while the hot forged one has a homogeneous more isotropic
 412fine-grained microstructure with almost no texture ([Figure 10](#)).



413

414Figure 9 - Diffraction pattern of samples with and without in-situ hot forging.

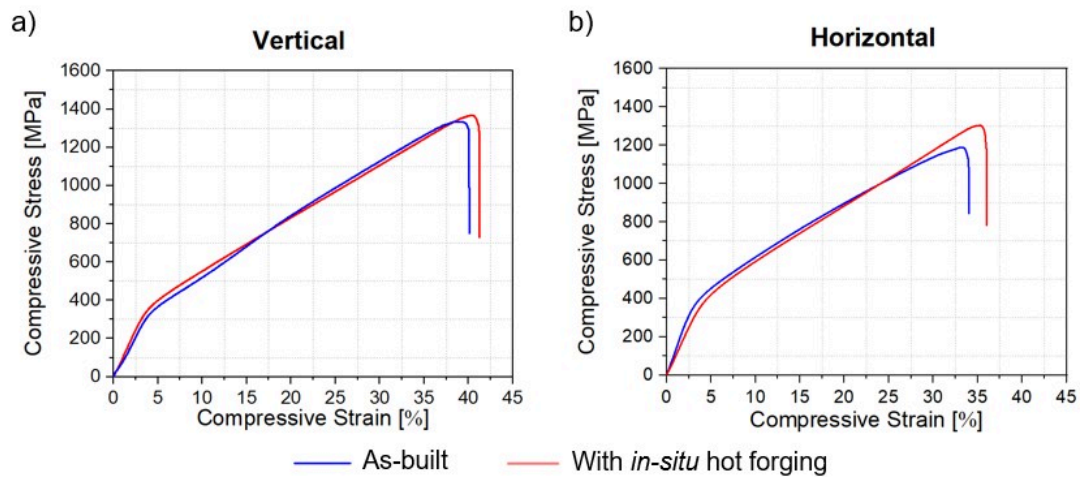
415



416
417 Figure 10 - 2D Debye-Scherrer patterns of samples: as-built (a) and in-situ hot forged (b).

418 To evaluate the influence of the *in-situ* hot forging on the mechanical properties and
419 anisotropy of the deposited material, uniaxial compression tests were preferred over
420 tensile ones, which allowed to evaluate specific zones in different directions using
421 reduced sized specimens.

422 The results of the mechanical tests performed are in agreement with the above
423 considerations on the microstructure. Representative compression stress-strain curves
424 for each sample obtained at different orientations are depicted in [Figure 11](#). The
425 summary of these compression tests is further detailed in [Table 3](#). It can be observed
426 that the use of *in-situ* hot forging during WAAM of the CuAl8 results in more isotropic
427 mechanical properties. In particular, the yield strength of the hot forged samples is
428 identical in the specimens obtained in the vertical and horizontal directions, while those
429 obtained without *in-situ* hot forging presented a higher yield strength in the horizontal
430 direction. The large and highly oriented grains in the CuAl8 deposits made without *in-*
431 *situ* hot forging justify this variation. The compressive strength of the parts was also
432 seen to show similar values for the vertical and horizontal *in-situ* hot forged CuAl8
433 WAAM parts, while for the as-built sample a difference of more than 100 MPa was
434 determined. These results further highlight the potential of *in-situ* hot forging WAAM in
435 contributing to a microstructure refinement by breaking the large columnar grain
436 structure typical of WAAM, which then results in more isotropic mechanical properties.
437 Fracture always occurred at approximately 45° with the applied force direction, as
438 expected for a ductile alloy as this one.



439
440 Figure 11 - Uniaxial compression stress-strain curves for the samples removed from horizontal a) and
441 vertical b) directions of the as-built and in-situ hot forged samples.

442

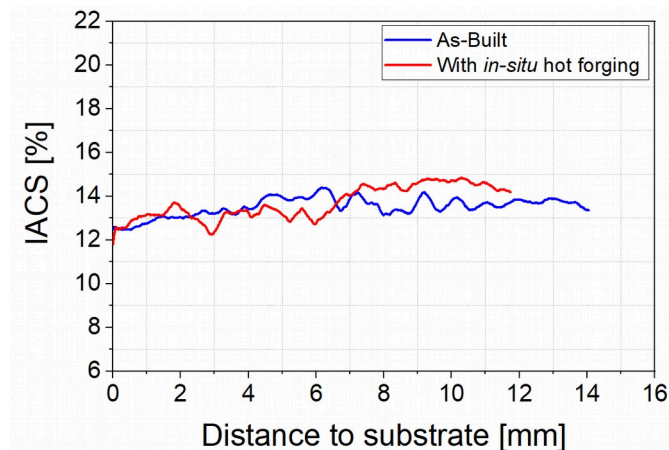
443 Table 3 - Summary of mechanical properties

Direction of sample removal		Yield strength [MPa]	Compressive strength [MPa]
Vertical	With hot forging	258 ± 33	1362 ± 64
	Without hot forging	245 ± 15	1334 ± 79
Horizontal	With hot forging	246 ± 13	1349 ± 42
	Without hot forging	292 ± 26	1216 ± 20

444

445 Since CuAl8 alloys are used in electromagnetic applications, the electrical conductivity
446 was measured to assess, whether the effect of the *in-situ* hot forging reduced this
447 property. [Figure 12](#) depicts the electrical conductivity profiles along the height of the
448 parts, and there is a negligible difference, showing that the *in-situ* hot forging WAAM
449 does not significantly affect the electric conductivity of this alloy, which can be
450 beneficial.

451 The fact that this alloy has an excellent electrical conductivity, prevents to have an
452 identification of the grain refined zones in the manufactured parts, as observed in other
453 alloys. Nevertheless, this method is sensitive to microstructural features as discussed
454 by Santos *et al.* [\[19\]](#), and it is evident that the as-built part has a coarser microstructure
455 due to more pronounced peaks and valleys.



456

457Figure 12 – Comparison of the electrical conductivity along the height of the parts with and without in-situ
 458hot forging. IACS means “International Annealed Copper Standard”, and it expresses the electric
 459conductivity as a percentage of the conductivity of pure annealed copper at 20° C. 100% IACS
 460corresponds to an electric conductivity of 58×10^6 S/m.

461

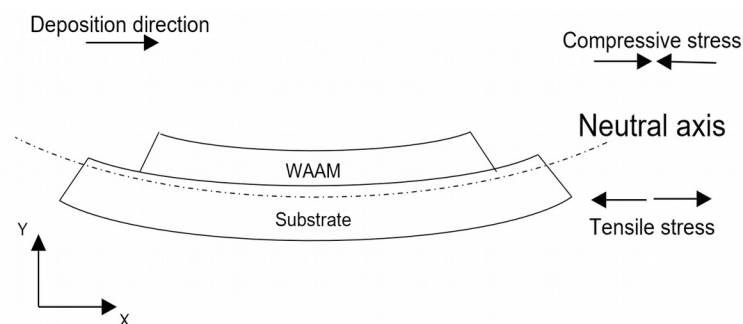
462Residual stress measurements

463In conventional WAAM parts, distortion builds up due to the accumulation of residual
 464stresses that are formed as a consequence of the thermal gradients during the
 465deposition process, with the same formation mechanism as those discussed in the arc
 466welding process [31]. To satisfy the equilibrium condition, the residual stresses present
 467on a part must always be balanced in such a way that the resultant force is null.
 468Moreover, the magnitude and direction of these stresses depend, among others, on the
 469material, the deposition strategy and parameter selection, as well as the dimensions
 470and geometry of the component.

471In WAAM, research most studies are performed on thin-walled samples, and the
 472analysis of residual stress is focused on three directions: longitudinal (X axis);
 473transverse (Z axis) and normal (Y axis), as illustrated in [Figure 13](#). Among these, the
 474directions where the development of residual stresses due to the thermal gradients is
 475more critical are the longitudinal and normal directions, since along the transverse
 476direction, the low width of the wall samples restricts the development of large residual
 477stresses.

478Additionally, regarding the longitudinal stresses measured along the building direction,
 479after unclamping the part from the baseplate, the specimen typically bent upwards to
 480balance the net bending moment across the section and achieve the equilibrium
 481condition, and tensile stress are observed at the bottom of the wall and steadily
 482decreased towards the top, where they tend show a compressive nature. Moreover, the

483 geometry of the substrate also influences the stress development on the deposited
 484 material. For depositions with few layers where the substrate accounts for a large
 485 amount of the total part height, the neutral axis can be positioned over the substrate,
 486 which puts all the deposited material under compressive residual stresses, as observed
 487 by Moat et al. [32] and is illustrated in Figure 13. While for higher samples where the
 488 substrate is a small percentage of the total part height, the neutral axis is positioned
 489 over the deposited material, and therefore, near the substrate the deposited material is
 490 under tensile residual stress and in the upper region it is under compressive stresses
 491 [33–35].



492

493 Figure 13 - Schematic representation of the longitudinal stress distribution in a WAAM part after
 494 unclamping.

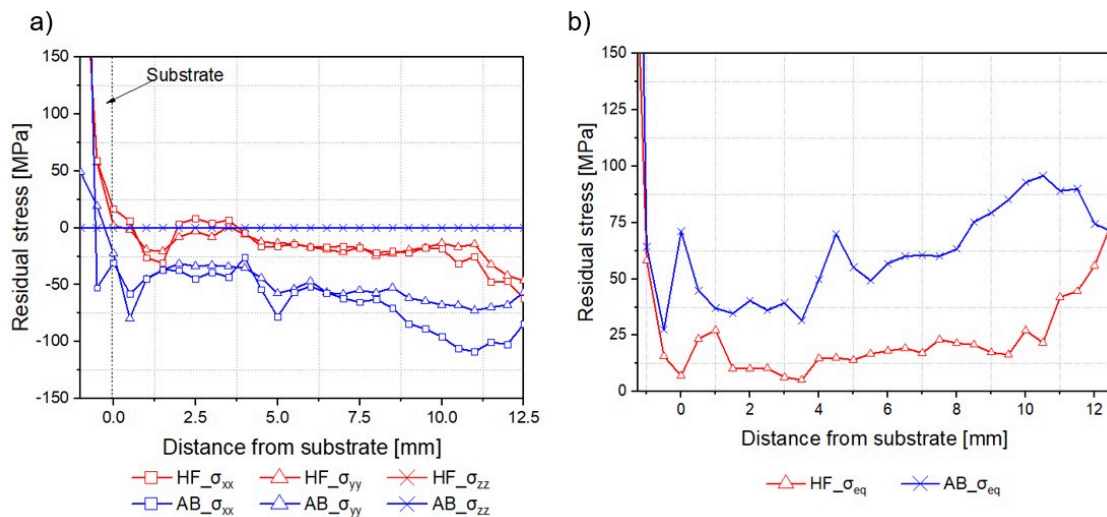
495 The directional and equivalent residual stress measurements as obtained from the high
 496 energy synchrotron X-ray diffraction measurements are shown in Figure 14. As
 497 expected, the transverse residual stress along the sample height is almost null in both
 498 samples. It is also observed that most of the longitudinal and normal tensile stresses
 499 are accommodated by the substrate, indicating that the neutral axis is situated near the
 500 interface between the substrate and the deposited material.

501 In the as-built sample, the stresses aligned with the longitudinal direction present a
 502 profile in agreement with the ones reported in the bibliography [35], where it is verified
 503 that from the bottom of the wall towards the top the stress steadily decreased.

504 It is clear that the *in-situ* hot forging reduces the magnitude of the residual stress
 505 formed due to thermal gradients in both longitudinal and normal directions. But the
 506 most relevant is the fact that the longitudinal stresses, instead of presenting a curvature
 507 with a negative slope, present an almost constant stress value along the height of the
 508 sample, which indicates that the material will have the same behavior regardless of the
 509 location where an external load may be applied. Moreover, the compressive residual
 510 stresses may also positively affect fatigue crack growth in both normal and longitudinal
 511 directions.

512 The variation in the magnitude of residual stresses due to forging occurs during the
 513 production of the part while the substrate is still clamped. When the material cools
 514 down to room temperature, after its deposition, the material wants to shrink due to the
 515 thermal contraction, but since the substrate is clamped, it limits this contraction of the
 516 material and therefore tensile stresses are developed. The *in-situ* hot forging occurs
 517 immediately after the deposition of the material, and the compressive stresses induced
 518 by the forging counteract the tensile stresses induced by thermal gradients. Thus, after
 519 unclamping the part, the stress redistribution needed to reach an equilibrium state is
 520 lower in the hot forged sample than in the as-built one, since the resultant magnitude of
 521 the residual tensile stresses is lower due to the positive effect induced by the
 522 compressive stress introduced by the hot forging.

523 Moreover, from a distance of 10 mm from substrate, a rapidly increase of the von
 524 Mises equivalent stress is observed in the *in-situ* hot forged sample, which is where the
 525 last layer starts, and thus the total compressive residual stresses formed during forging
 526 are still present since there is no subsequent layer to partially remelt the forged
 527 material. This is corroborated by the measurements of the principal stresses showed in
 528 [Figure 14 a\)](#), where an increase of the compressive stresses is observed in the hot
 529 forged sample, particularly in the normal (Y axis) and longitudinal (X axis) directions, in
 530 contrast to the as-built sample.



531

532 Figure 14 - Residual stresses measured by synchrotron X-ray diffraction in the as-built and *in-situ* hot
 533 forged samples: a) principal stresses; b) von Mises equivalent stress. The “HF” red curve corresponds to
 534 the *in-situ* hot forged condition while “AB” blue curve corresponds to the as-built condition.

535

536

537

538

539 **6. Conclusions**

540 This paper presents the results of a study performed in *in-situ* hot forging wire and arc
541 additive manufacturing of a CuAl8 alloy with a circular hammer to improve the forged
542 area, while allowing to produce part shapes other than linear. A dynamic study of the
543 process is presented analyzing the forces applied in the visco-plastic regime. The
544 following major conclusions were drawn:

545- The CuAl8 alloy does not develop defects during arc-based additive manufacturing
546 with or without the *in-situ* hot forging variant;

547- the application of *in-situ* hot forging clearly refines and homogenizes the
548 microstructure without significantly affecting the existing phases. Qualitatively, the hot
549 forging operation seems to slightly increase the β phase fraction;

550- since this is a highly ductile alloy there is no significant variation in hardness;

551- compressive strength measured showed a more homogeneous resistance in vertical
552 and horizontal directions when the part is hot forged, while the as-built part has a more
553 significant difference in both directions. As expected, fracture occurred at
554 approximately 45°;

555- hot forging does not affect the electrical conductivity of WAAM parts;

556- hot forging reduces and homogenize the residual stress of parts.

557 **Acknowledgments**

558 Authors acknowledge the Portuguese Fundação para a Ciência e a Tecnologia (FCT -
559 MCTES) for its financial support via the project UID/EMS/00667/2019 (UNIDEMI). VD
560 acknowledges FCT - MCTES for funding the PhD grant SFRH/BD/139454/2018. TAR
561 acknowledges FCT - MCTES for funding the PhD grant SFRH/BD/144202/2019.
562 Funding of CENIMAT/i3N by national funds through the FCT-Fundação para a Ciência
563 e a Tecnologia, I.P., within the scope of Multiannual Financing of R&D Units, reference
564 UIDB/50025/2020-2023 is also acknowledge. This activity has received funding from
565 the European Institute of Innovation and Technology (EIT) Raw Materials through the

566project Smart WAAM: Microstructural Engineering and Integrated Non-Destructive
 567Testing. This body of the European Union receives support from the European Union's
 568Horizon 2020 research and innovation programme. Parts of this research were carried
 569out at PETRA III at DESY, a member of the Helmholtz Association. The research
 570leading to this result has been supported by the project CALIPSOplus under the Grant
 571Agreement 730872 from the EU Framework Programme for Research and Innovation
 572HORIZON 2020. This project has received funding from the EU-H2020 research and
 573innovation programme under grant agreement No 654360 having benefitted from the
 574access provided by PETRA III at DESY in Hamburg, Germany within the framework of
 575the NFFA-Europe Transnational Access Activity. The authors acknowledge support by
 576OCAS NV and GUARENTEED via Joachim Antonissen.

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717Remark: The supplementary material is temporarily available in the Drive folder here:

718<https://drive.google.com/drive/folders/1SFFlhJmL5p3IkQis8cB6UVWva3wozGi?usp=sharing>